



Experimental Probe of Inflationary Cosmology (EPIC)

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CMB: The Most Powerful Tool in Cosmology

... Which Remains Largely Unexplored

Relic from the Early Universe

Discovery by Penzias & Wilson

- A revolution in cosmology
- Strong evidence for Big Bang model

BB Spectrum & anisotropies by COBE

- Big Bang model unassailable
- Support for Inflation

Precision Cosmology

Discovery of first acoustic peak

- Flat universe – confirms Inflation

Temperature power spectrum, WMAP

- Energy/matter content of Universe

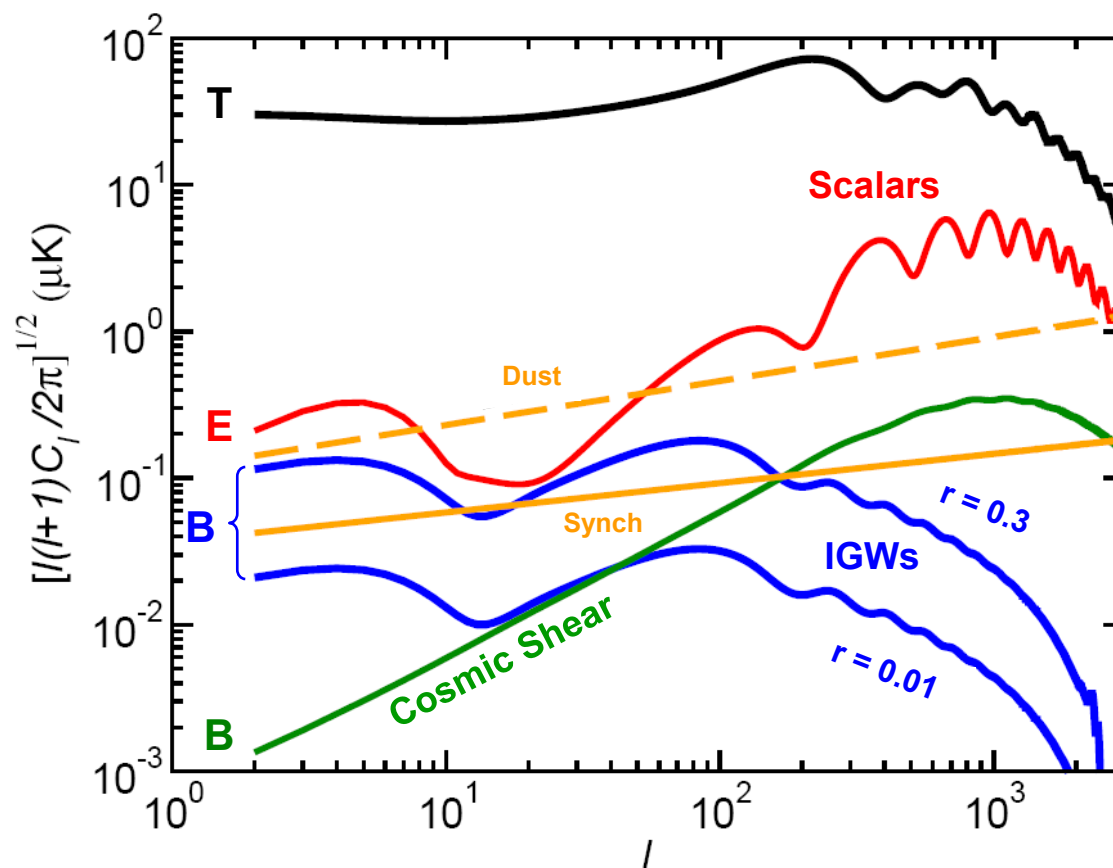
Enter Polarization

Gravitational Wave Polarization

- Best probe of Inflationary physics

A Wealth of Cosmology

- Reionization, Gravitational Lensing





EPIC is a Scan-Imaging Polarimeter



Boomerang

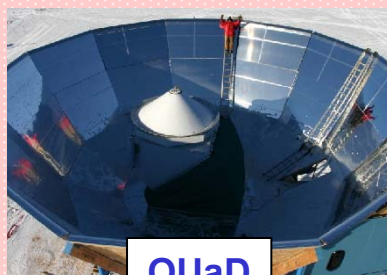


Maxipol

PAST



BICEP

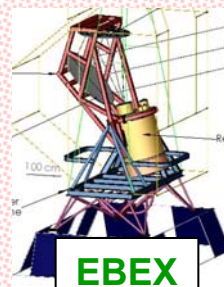


QUaD

ACTIVE



Planck



EBEX

PLANNED



Polarbear



Spider

FUTURE

Scan detectors across sky to build CMB map

Simple technique. Established history in CMB.

Scaling to higher sensitivity → Better arrays

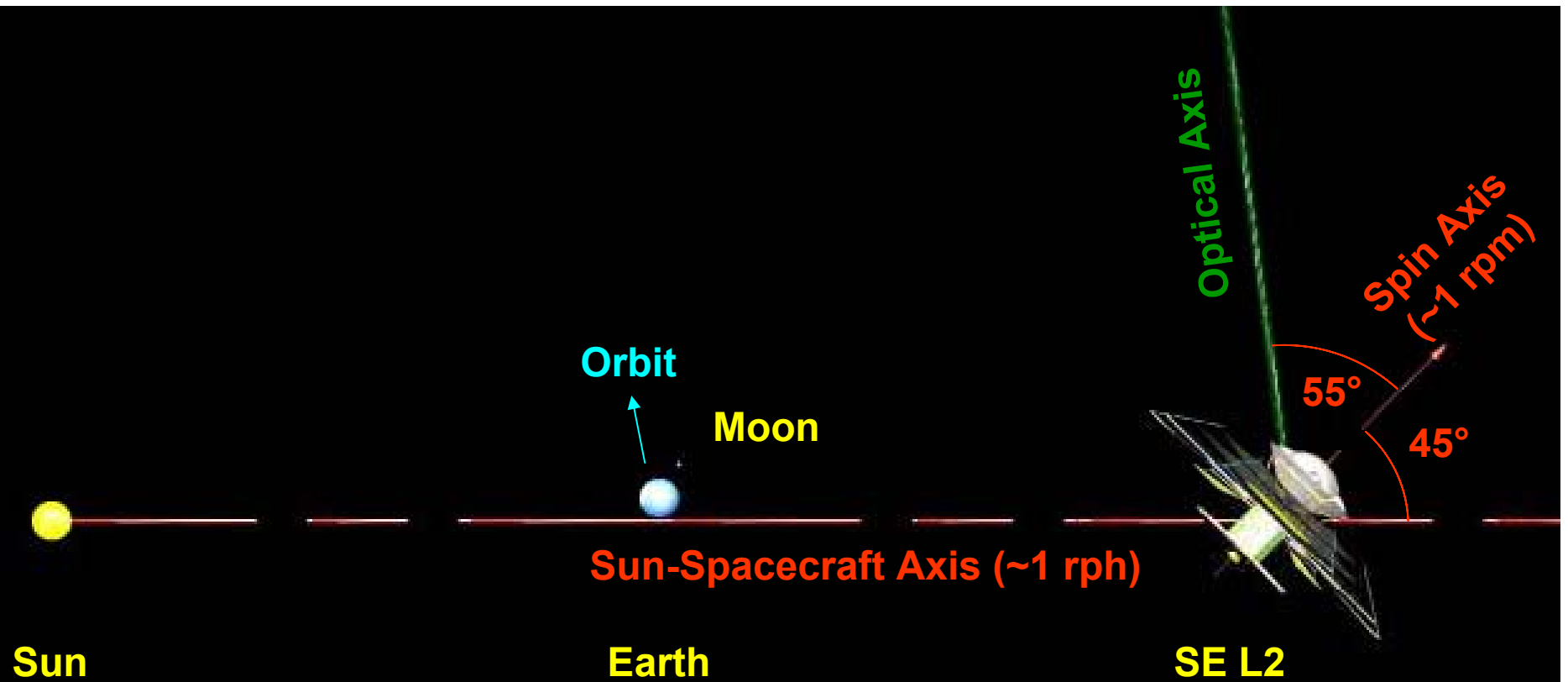
Adapt this technique for precision polarimetry

EPIC mission concept

High-TRL, simple, & flexible architecture

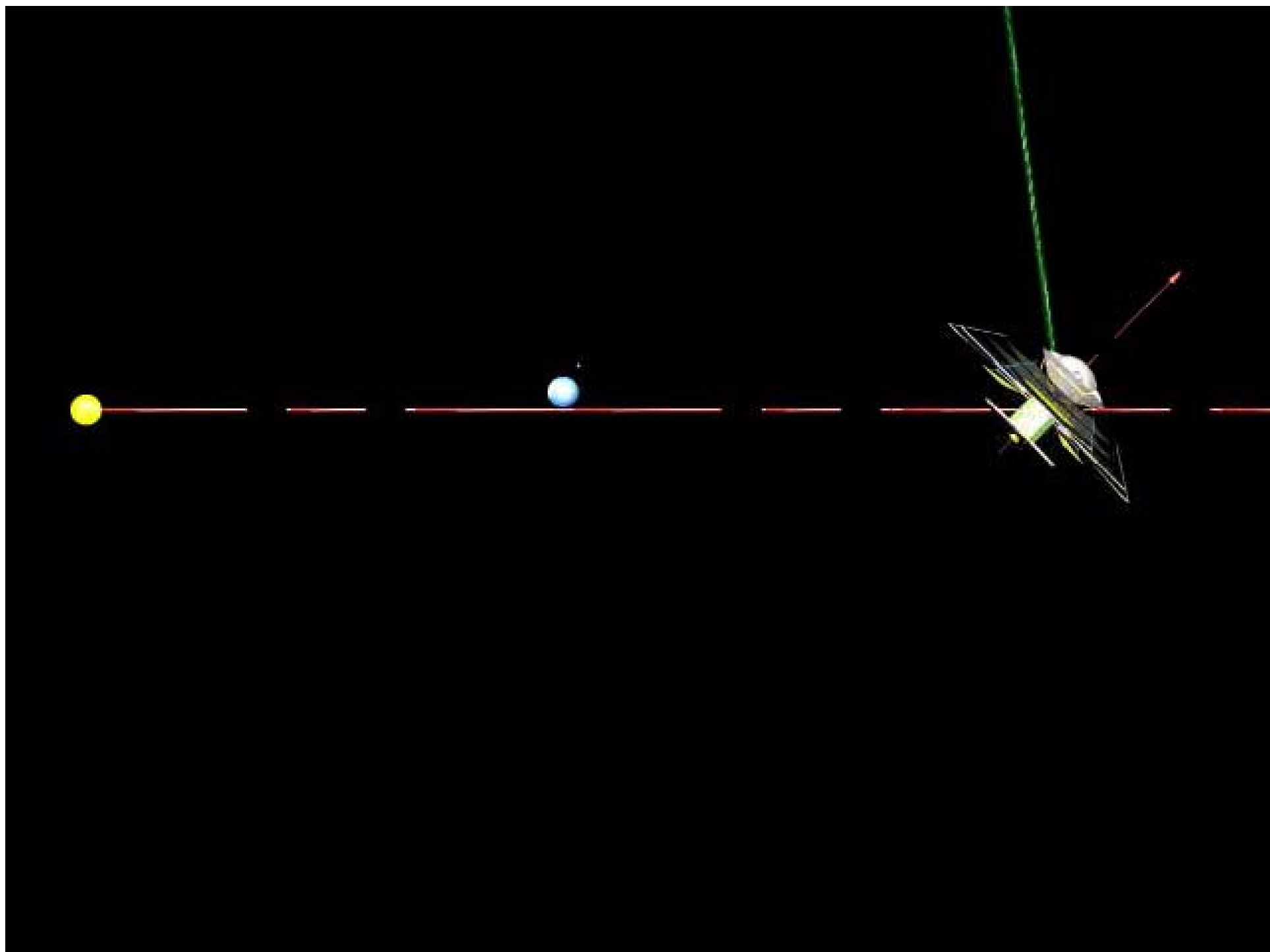
High sensitivity, wide band coverage, low angular resolution*

*See Weiss Committee TFCR Report: astro-ph 0604101



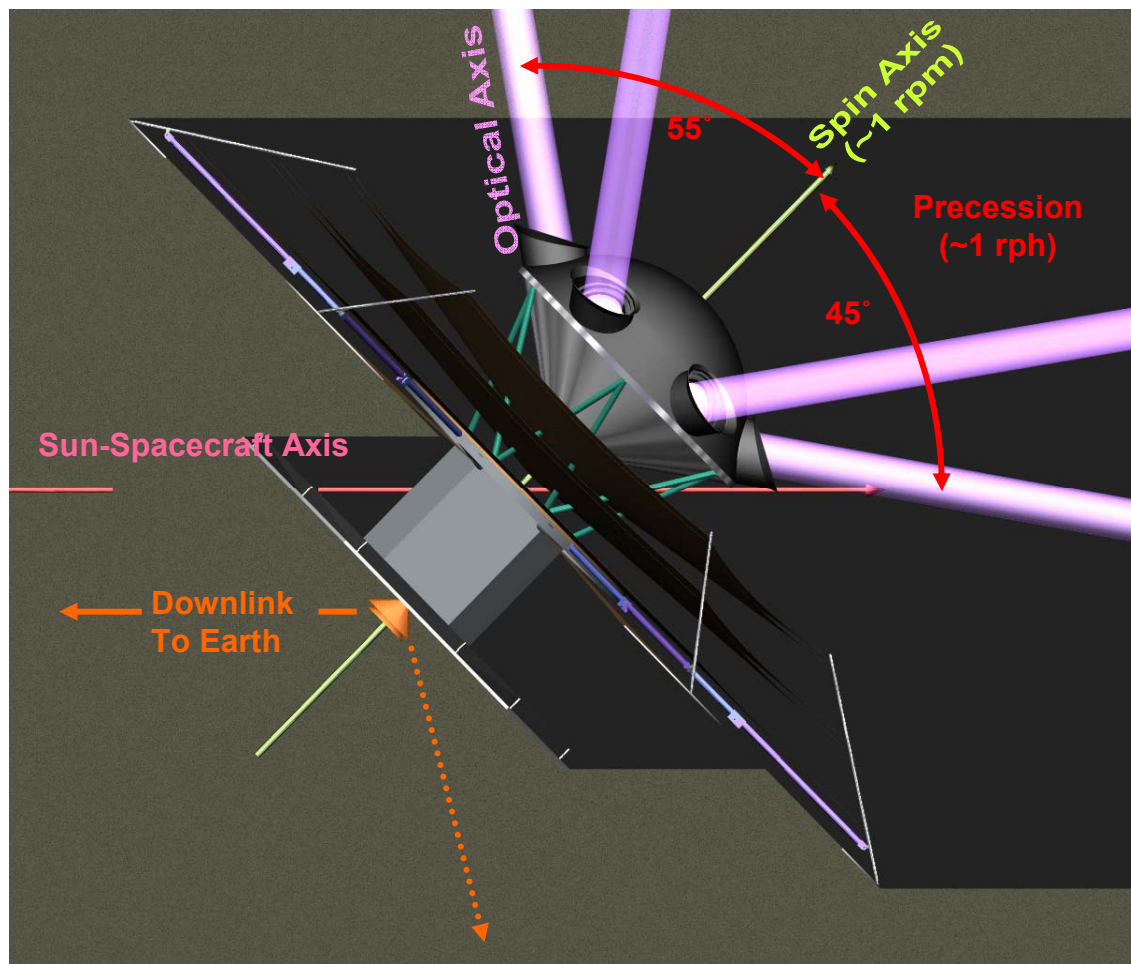
Why Space?

- All-Sky Coverage
- High Sensitivity
- Systematic Error Control
- Broad Frequency Coverage



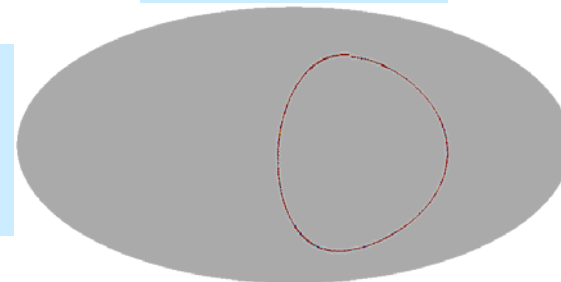


Scan Strategy

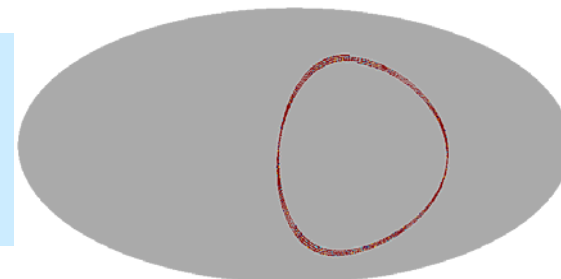


Scan Coverage

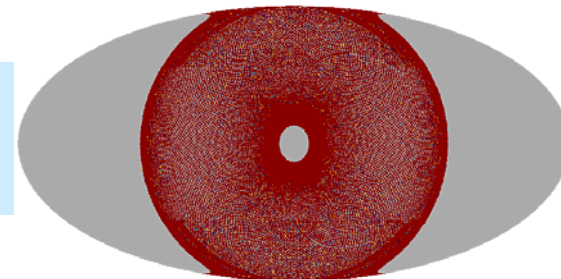
1 minute



3 minutes

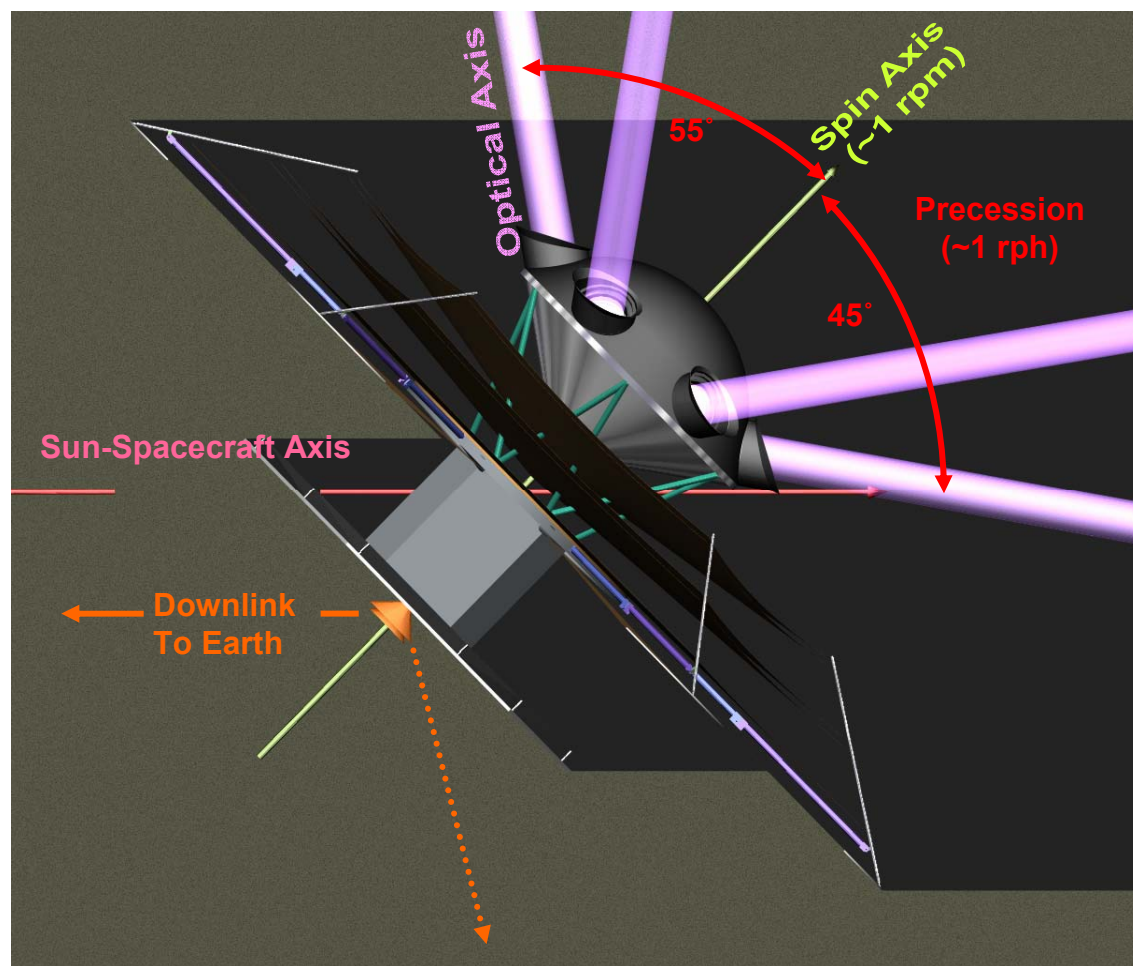


1 hour

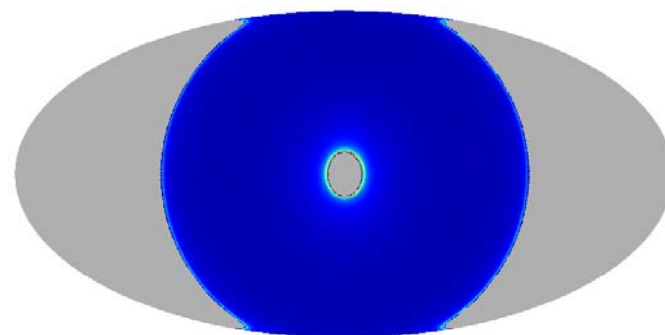




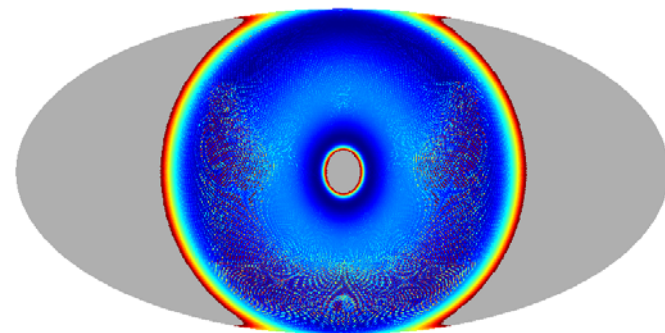
Scan Strategy



1 Day Maps



Spatial Coverage



Angular Uniformity

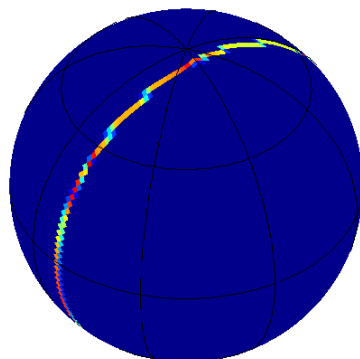
More than half the sky in a single day!



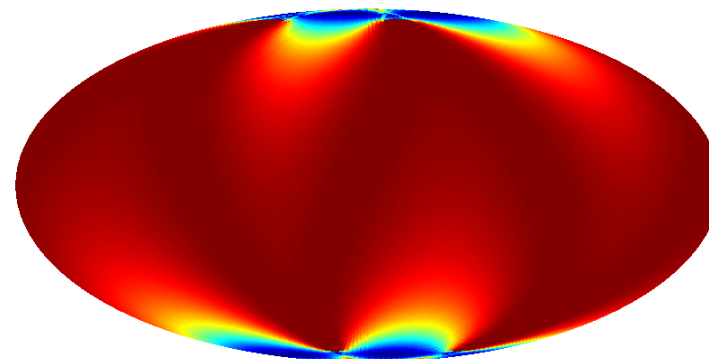
Redundant & Uniform Scan Coverage

Planck

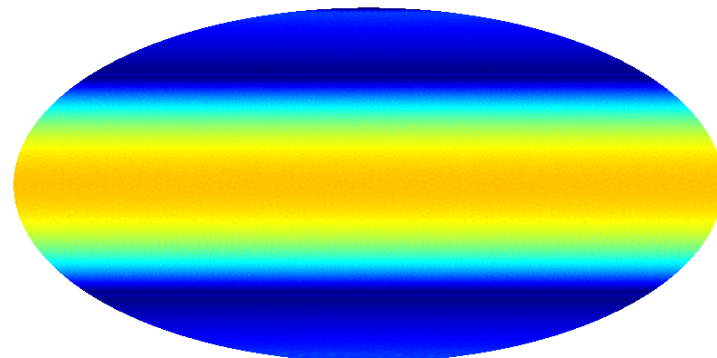
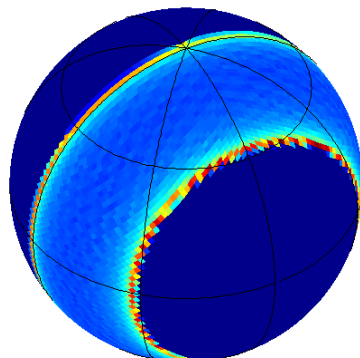
N-hits (1-day)



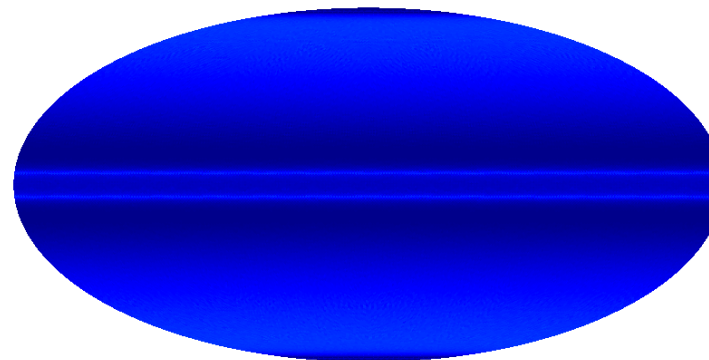
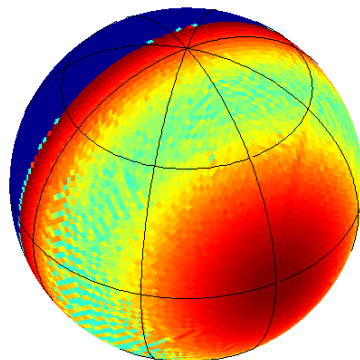
Angular Uniformity* (6-months)



WMAP



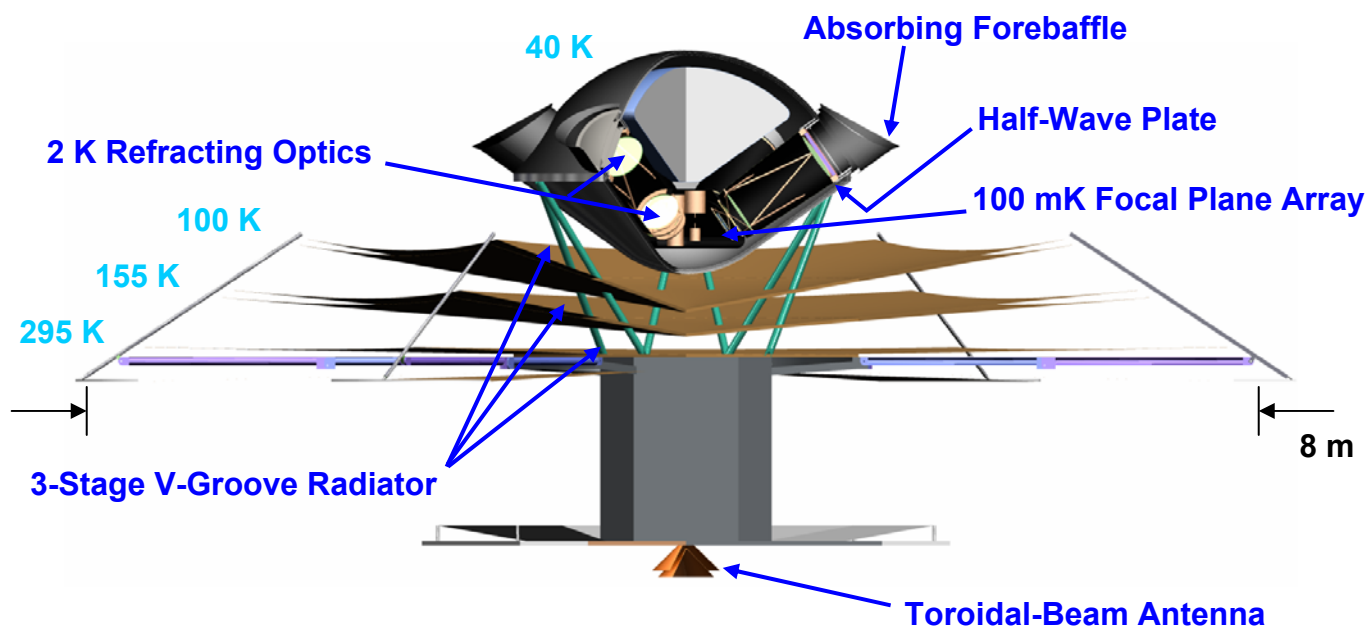
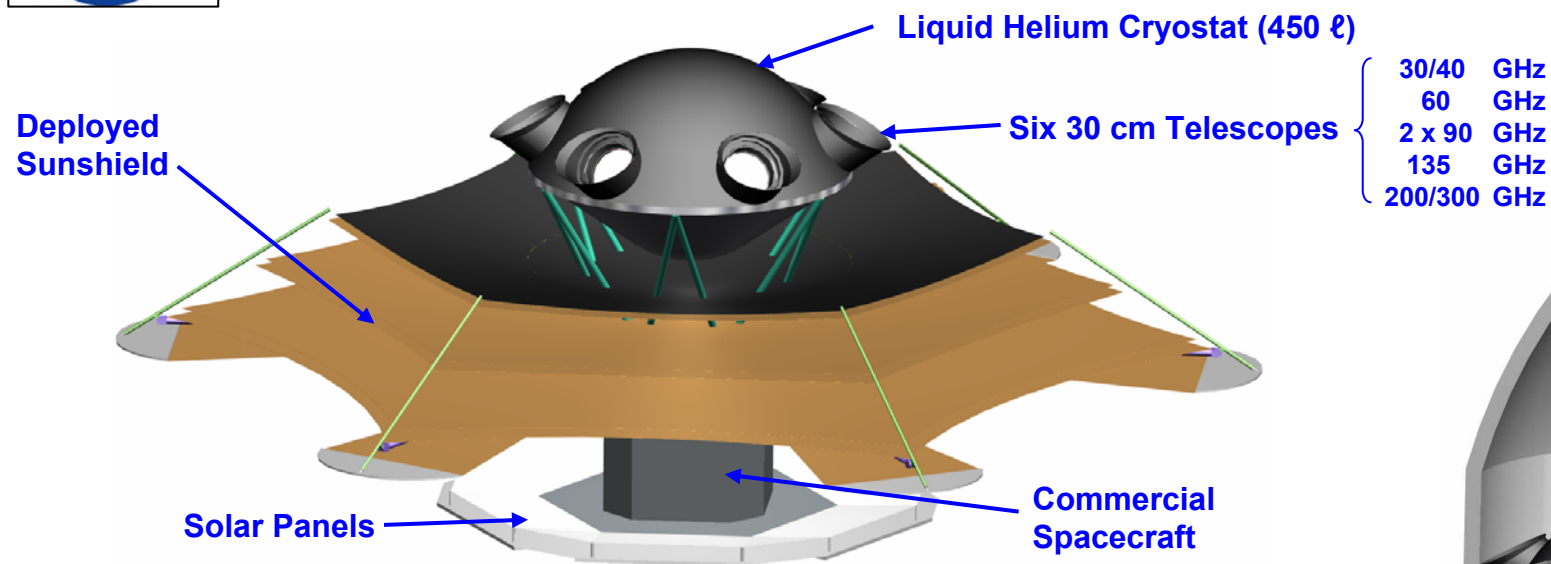
EPIC



* $\langle \cos 2\beta \rangle^2 + \langle \sin 2\beta \rangle^2$ 0  1



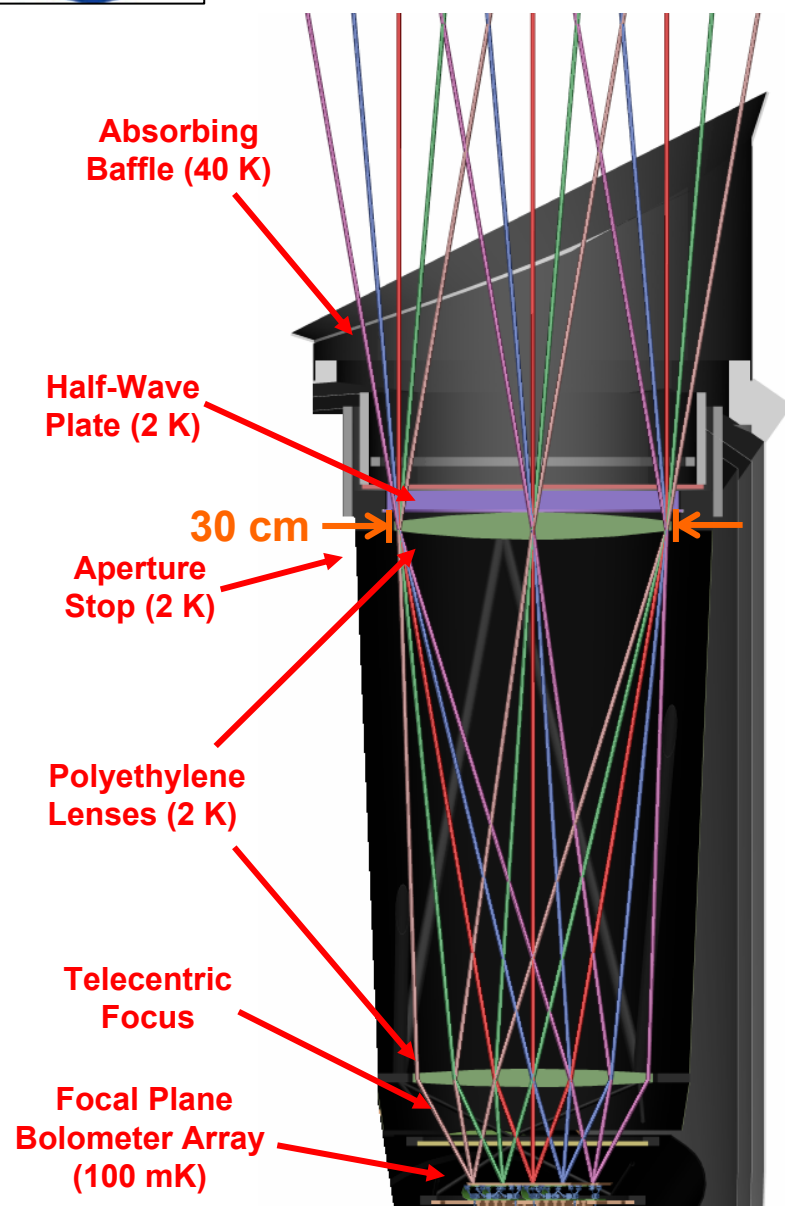
EPIC Mission Architecture



Delta 2925H 3-m



High Sensitivity with High-TRL Focal Planes



EPIC Projected Bands and Sensitivities

Freq [GHz]	FWHM [arcmin]	NTD Ge Bolometers			Planck
		# [det's]	NET/det* [$\mu\text{K}\sqrt{\text{s}}$]	NET* [$\mu\text{K}\sqrt{\text{s}}$]	NET/det† [$\mu\text{K}\sqrt{\text{s}}$]
30	155	8	69	24.5	
40	116	54	60	8.2	
60	77	128	49	4.4	
90	52	256	42	2.6	102
135	34	256	38	2.4	83
200	23	64	41	5.1	134
300	16	64	95	11.8	404
Total		830		1.5	

*Design Sensitivity †Goal Sensitivity

Note modest detector improvement over Planck due to relaxed time constant specification & 2 K optics

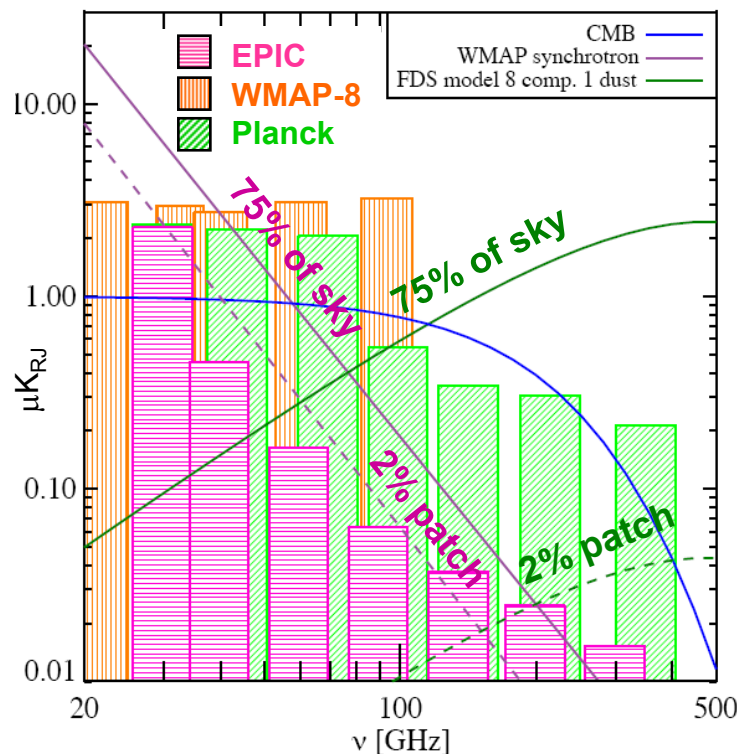
Planck Projected Sensitivities

Instrument	Technology	NET [$\mu\text{K}\sqrt{\text{s}}$]	Value
HFI 100 – 850 GHz	NTD Ge Bolometer	20.3	Design Goal
		< 16.7	Measured
LFI 30 – 70 GHz	HEMT	68.9	Req't

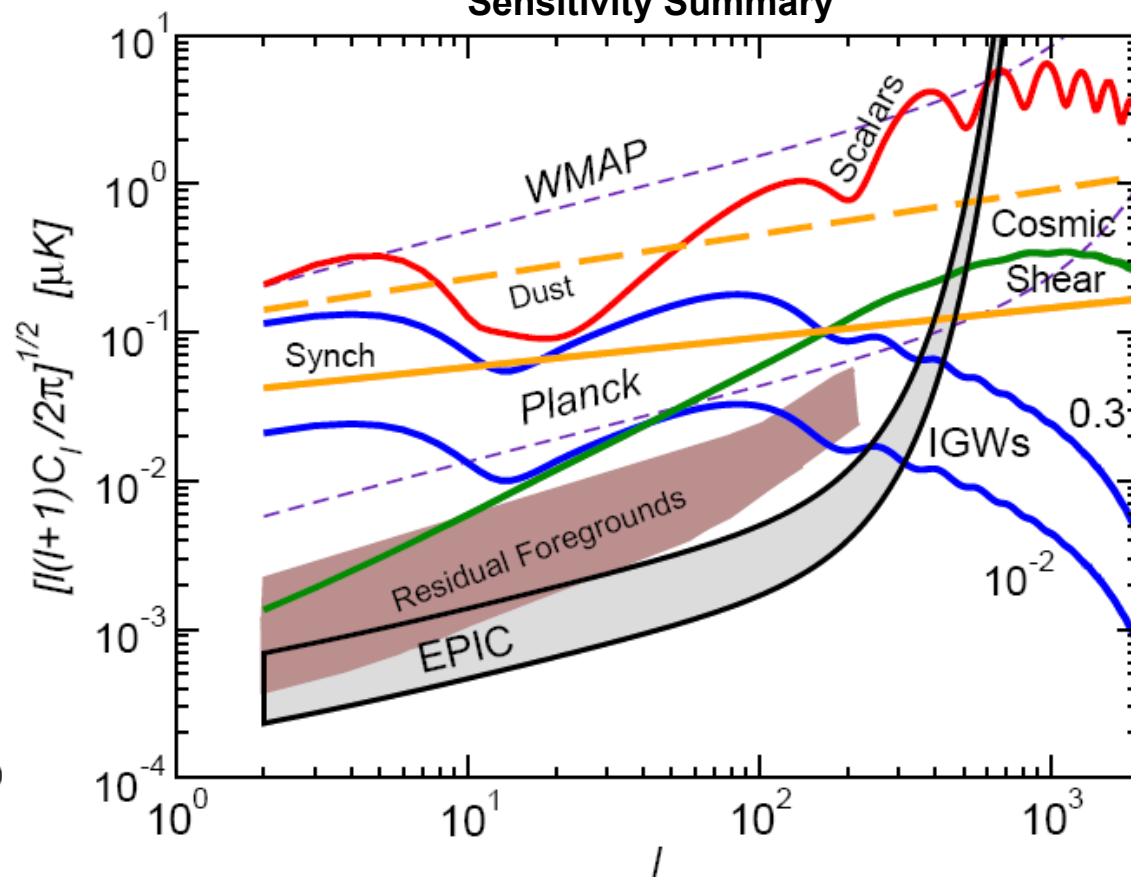


Measuring B-Polarization to Lensing Limit

Multi-Band Foreground Monitoring



Sensitivity Summary



1. Input Sky Model

- Synchrotron
- Thermal dust
- Free-free
- Spinning dust

2. Multi-band Removal

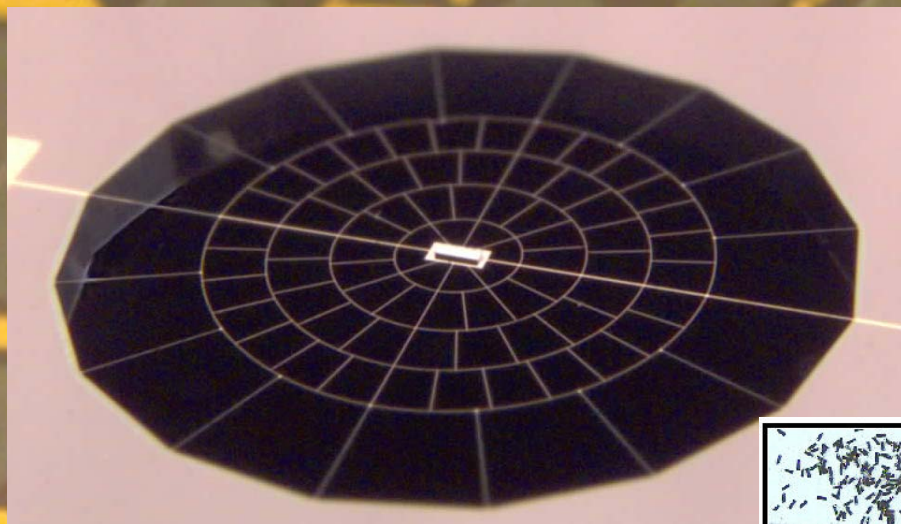
- Fit components spectrally
- Let indices float spatially

3. Results

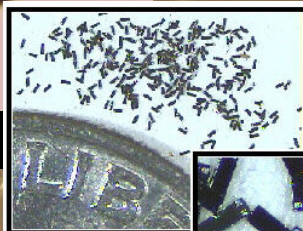
- Removal gives a *modest* sensitivity loss
- Better sensitivity gives better subtraction



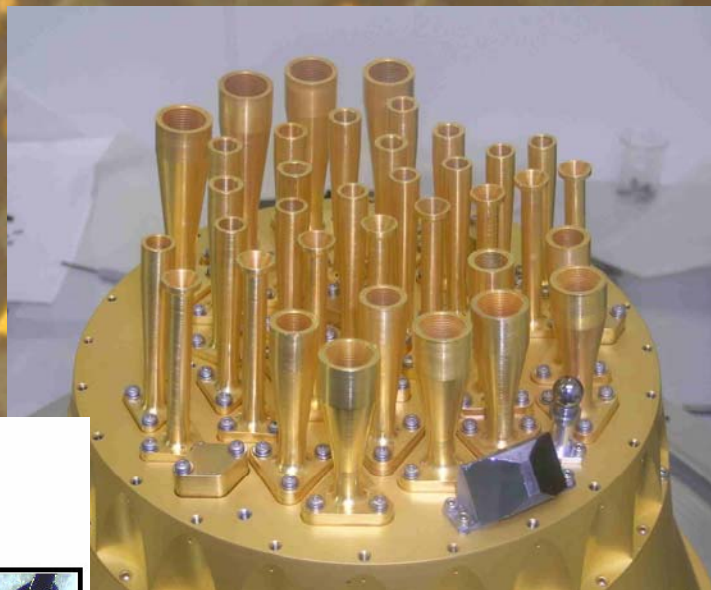
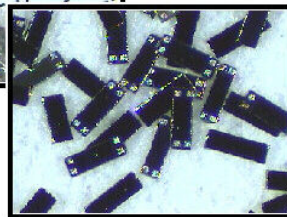
NTD Bolometers for Planck & Herschel



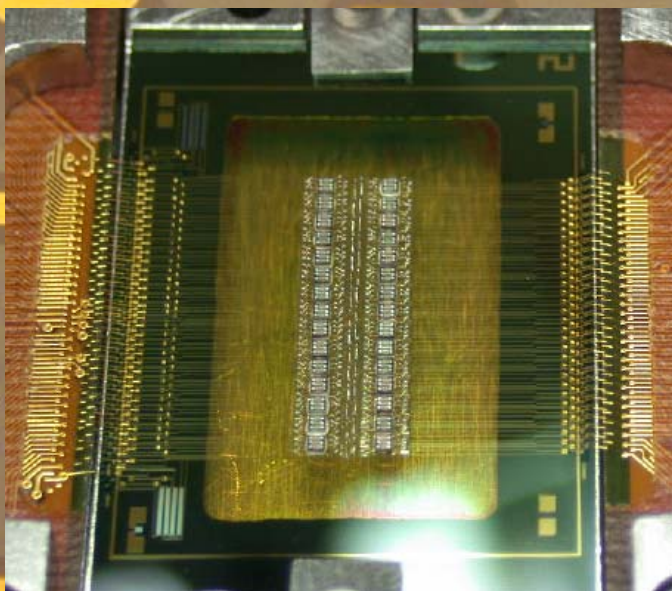
143 GHz Spider-web Bolometer



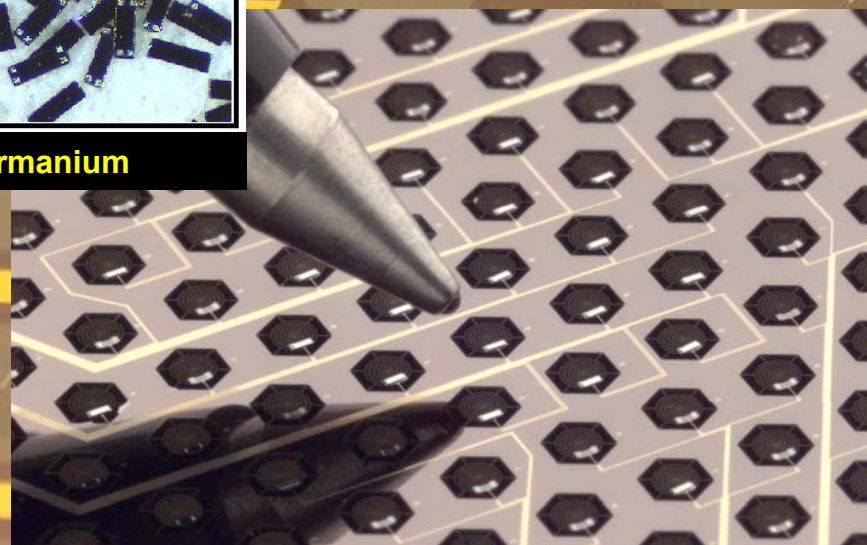
NTD Germanium



Planck/HFI focal plane (52 bolometers)



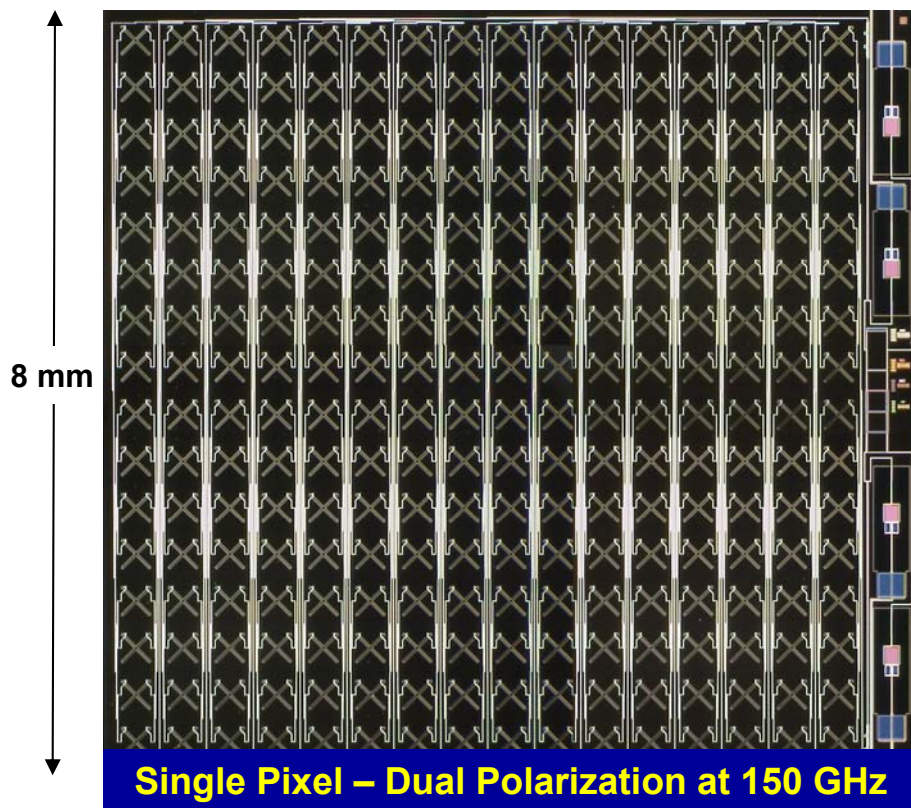
SPIRE Low-power JFETs



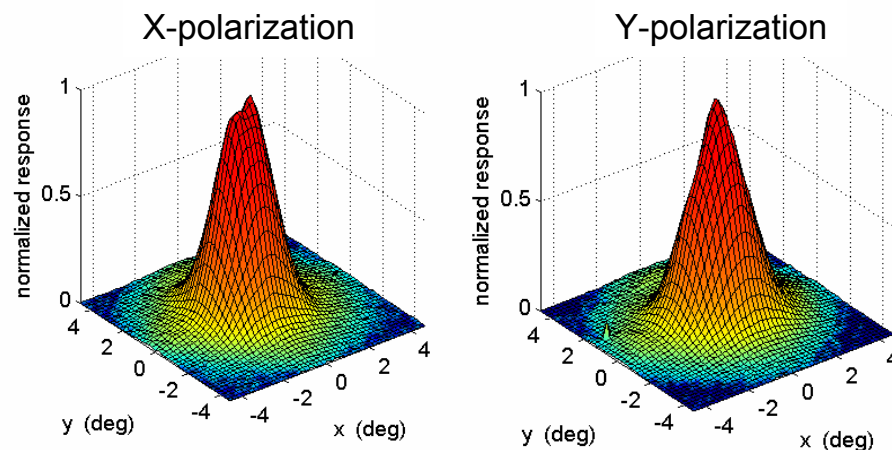
Herschel/SPIRE Bolometer Array



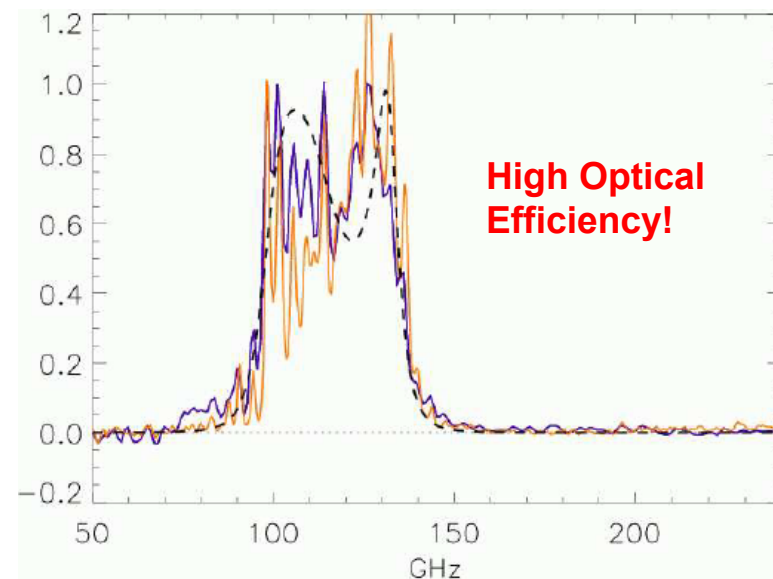
Antenna-Coupled Bolometers for EPIC



Measured Beam Patterns



Measured Spectral Response



Advantages

Small active volume..	30 – 300 GHz operation
Beam collimation.....	Eliminates discrete feeds
	Reduced focal plane mass
Intrinsic filters.....	No discrete components



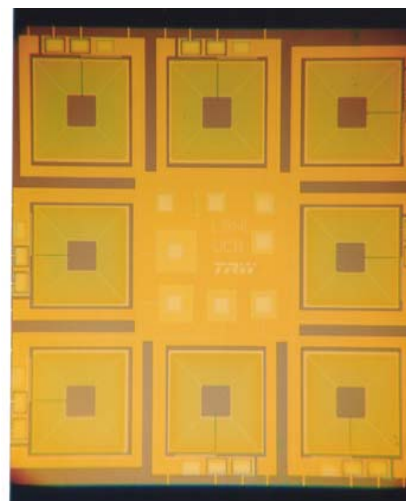
TES Bolometer Arrays

Advantages

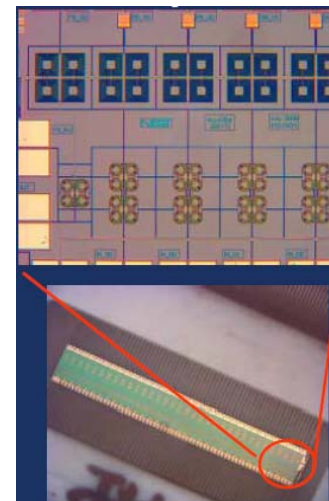
Multiplexing..... Larger array formats
Higher sensitivity
Fewer wires to 100 mK
Low cryogenic power disp.

Faster response... Useful for larger aperture

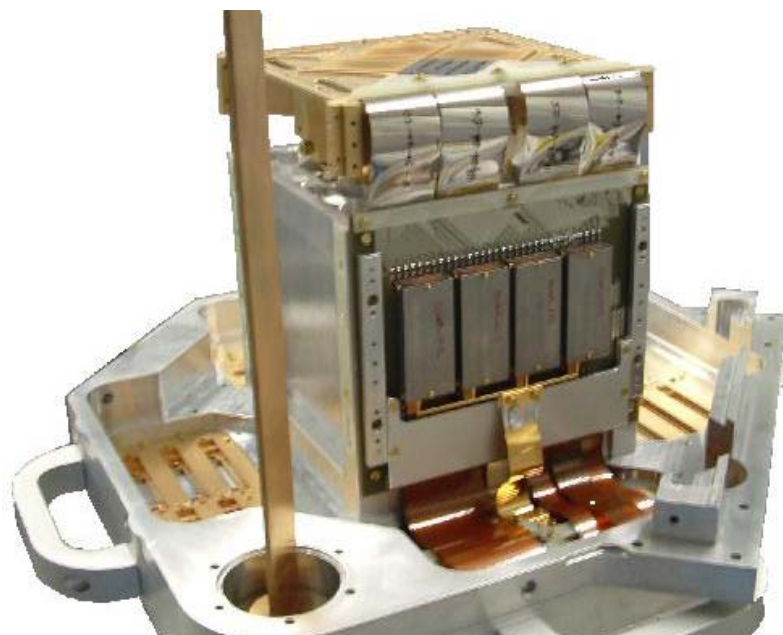
SQUID Multiplexing



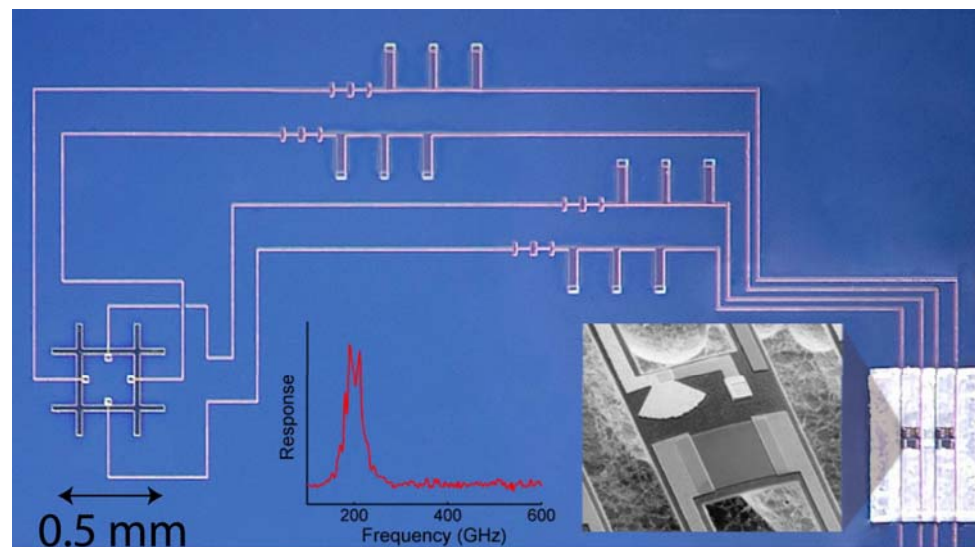
Freq. Domain (UCB/LBNL)



Time Domain (NIST)



SCUBA2 Focal Plane (10,000 TES Bolometers)



Antenna-Coupled TES Bolometers (UCB/LBNL)



EPIC Employs High-TRL Technologies

Technology	TRL	Heritage
Focal Plane Arrays (NTD Ge bolometers)		
NTD thermistors and readouts	8	Planck & Herschel
Antennas	4	
Wide-Field Refractor	6	BICEP
Waveplate (stepped every 24 hours)		
Optical configuration	6	SCUBA, HERTZ, etc.
Cryogenic stepper drive	9	Spitzer
LHe Cryostat	9	Spitzer, ISO, Herschel
Sub-K Cooler: Single-shot ADR	9	ASTRO-E2 (single-shot)
Deployable Sunshield	4-5	TRL=9 components
Toroidal-Beam Downlink Antenna	4-5	TRL=9 components

New technologies bring enhanced capabilities...

Antenna-Coupled TES / MKID detectors	→	Higher sensitivity, less mass & power
Continuously rotating waveplate	→	Better beam control
Continuous ADR	→	Less mass, no interruptions

...but we have the technology to do this mission today.

We need technology *evolution* not *revolution*



Systematic Error Mitigation Strategy

Instrument Design Goal	Instrument Requirement
Suppress <i>raw</i> systematic effect below statistical noise level	Suppress systematic errors below $r = 0.01$ after correction

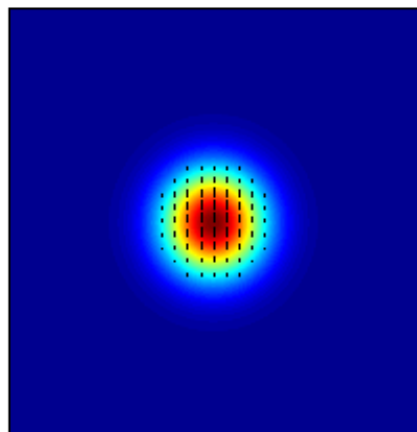
Systematic Effect	Design Goal	Mitigations	Heritage
Main Beam Effects	Δ Beamsize	1e-4	BICEP [†] SPIDER [‡] ---
	Δ Gain	8e-5	
	Δ Beam Offset	5e-5	
	Δ Ellipticity	1e-3	
	Pol. Beam Rot'n	3e-4	
Polarized Sidelobes	1 nKrms	• Refractor + baffle	BICEP [*]
1/f Noise	16 mHz	• Drift-scanned NTD bolometers	BOOMERANG [*]
		• (TES + modulator)	EBEX/SPIDER/ POLARBEAR [‡]
Scan-Synch Signals	1 nKrms	• Scan crossings	---
Optics Temp. Stability	10 μ Krms s/s 300 μ K/ $\sqrt{\text{Hz}}$	• Scan redundancy	---
Focal Plane Temp. Stability	3nKrms s/s 150 nK/ $\sqrt{\text{Hz}}$	• Dual analyzers	Planck [*]
		• Temperature monitoring & control of all critical stages	Planck [*]

* = Already demonstrated to EPIC requirement
 † = Proof of operation but needs improvement
 ‡ = Planned demonstration to EPIC requirement

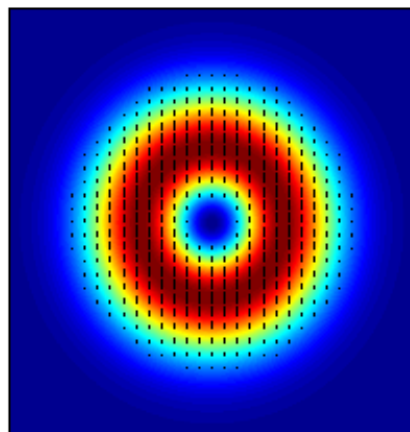


Polarization Systematics: Main Beams

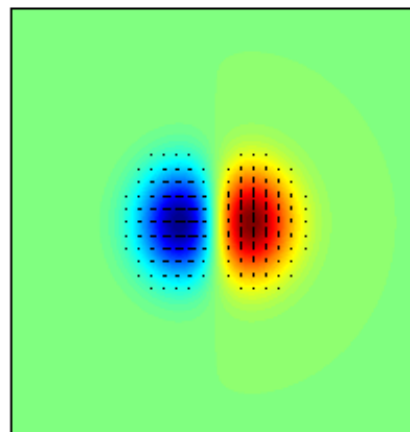
Main Beam Instrumental Polarization Effects



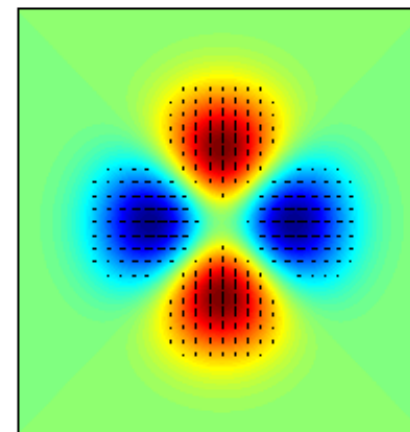
Differential Gain, Rotation



Differential FWHM



Differential Beam Offset



Differential Ellipticity

Calculation

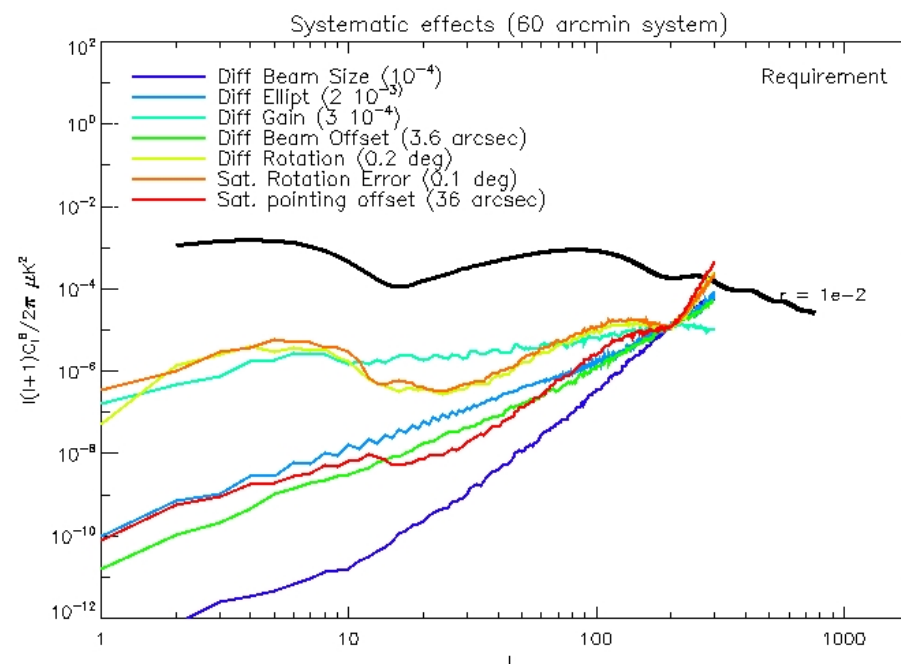
- Difference PSB beam pairs
- Parameterize main beam effects
- Convolve w/ scan pattern
- Calculate resulting power spectrum

Result

- Calculated telescope performance meets or exceeds requirements

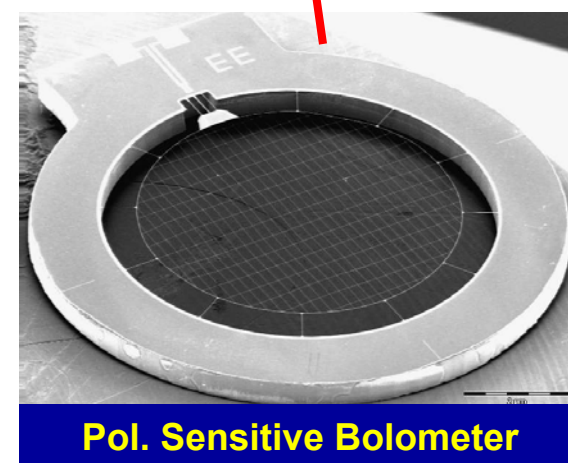
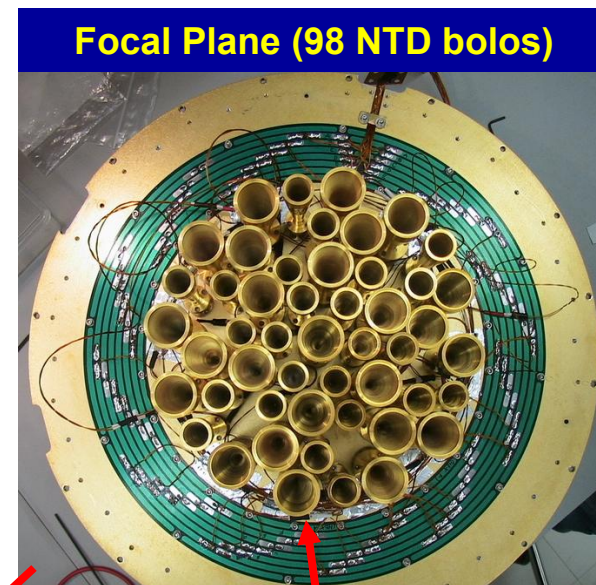
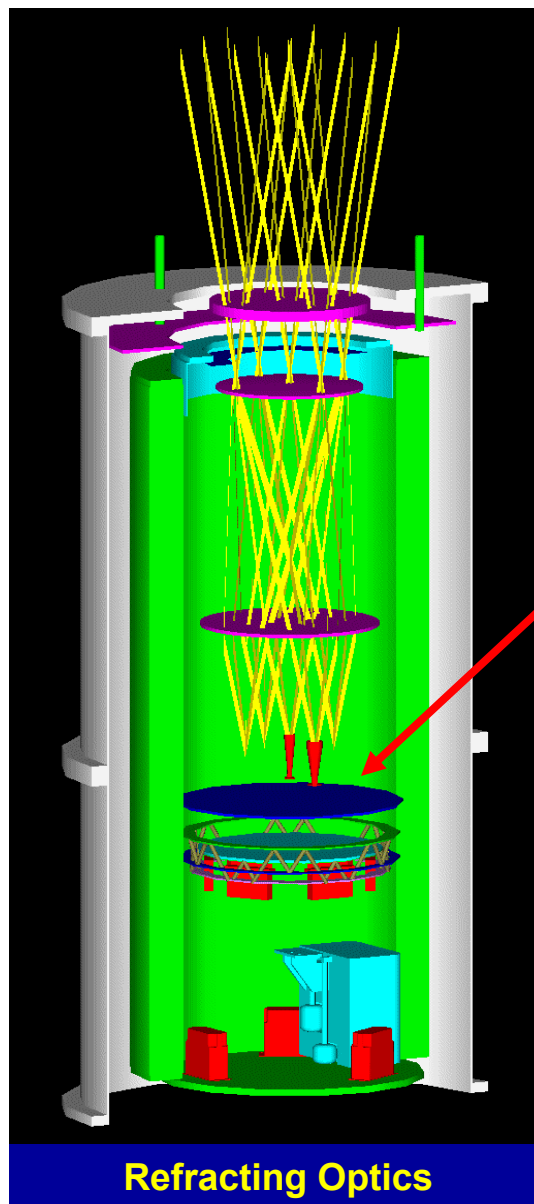
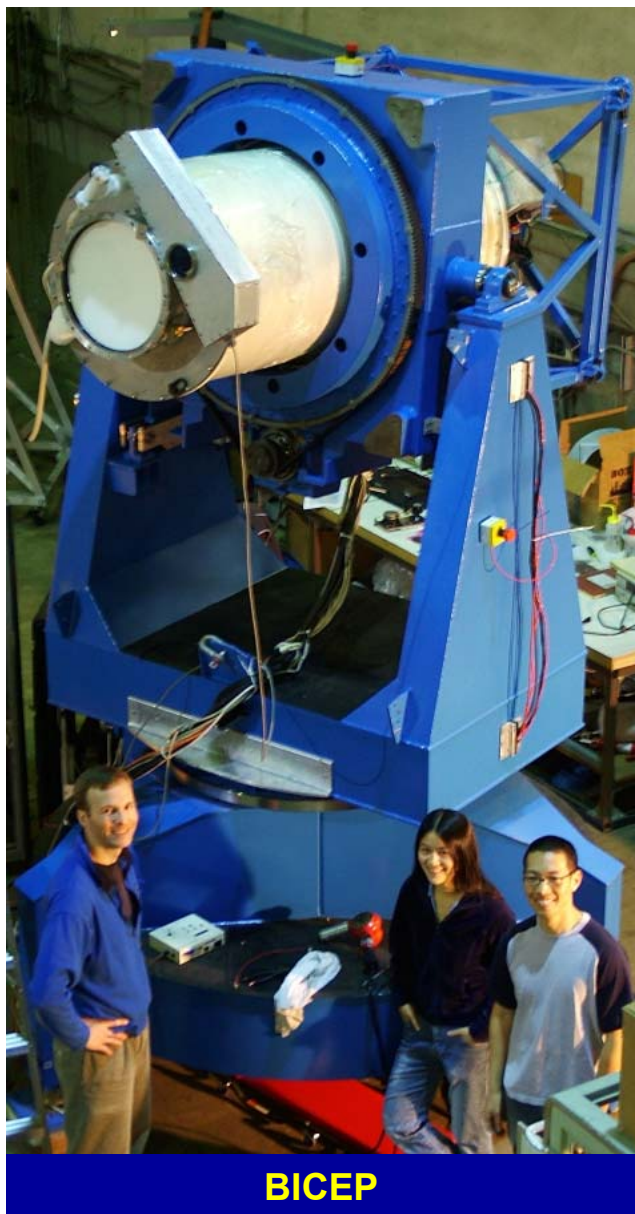
Caveats

- Calculation is conservative
- We can always apply *post facto* correction
- Waveplate virtually *eliminates* these effects





Demonstration of EPIC Technologies



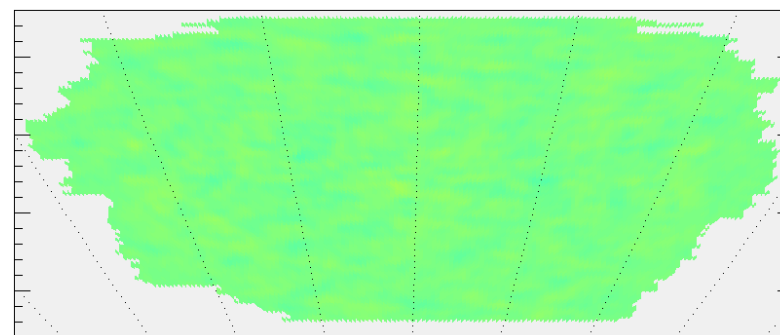
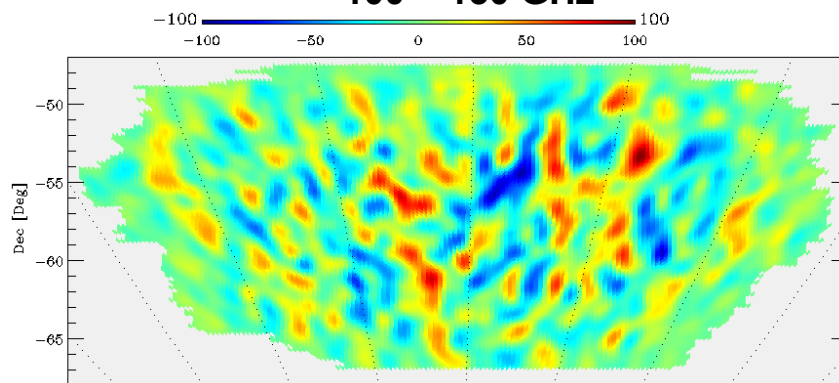


First-Season Maps from BICEP

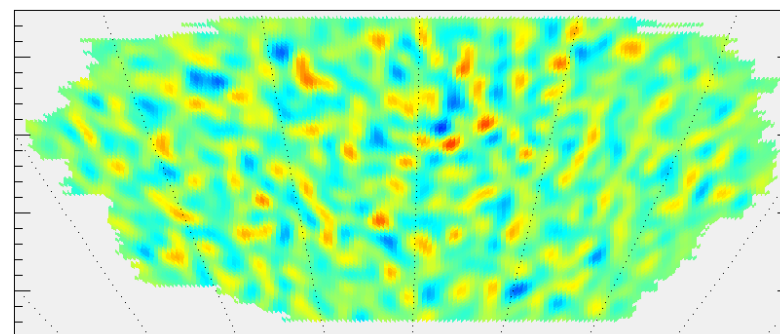
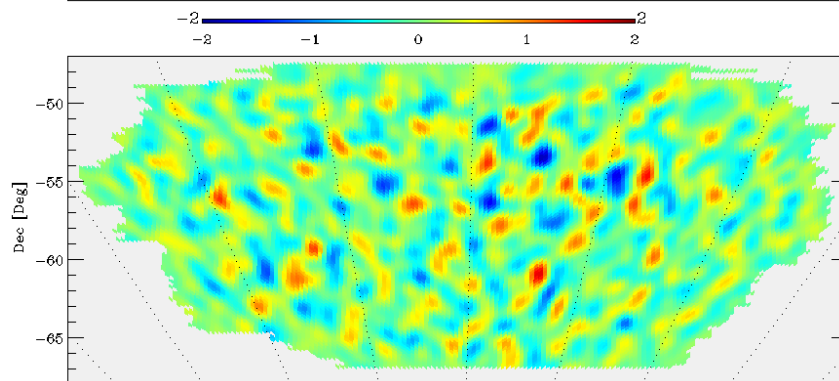
Temperature

100 + 150 GHz

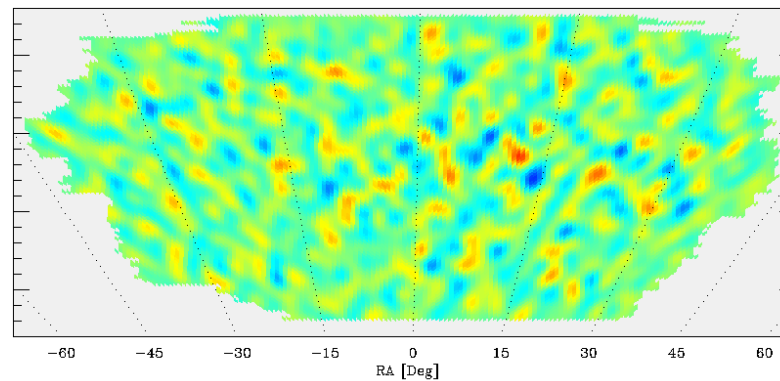
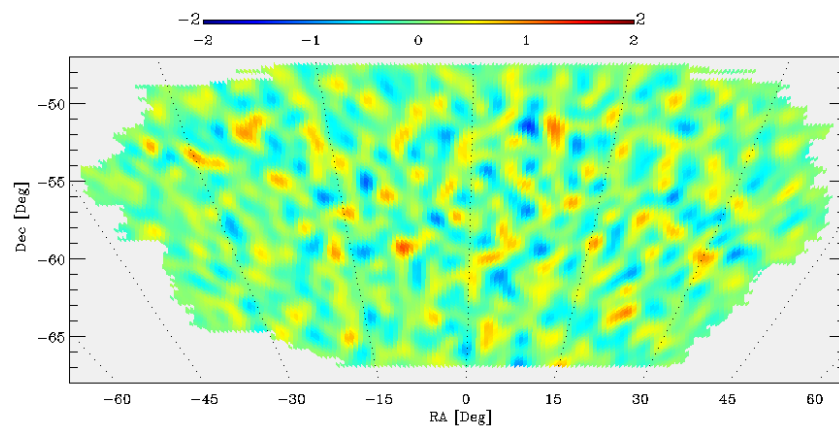
100 - 150 GHz



E-Polarization



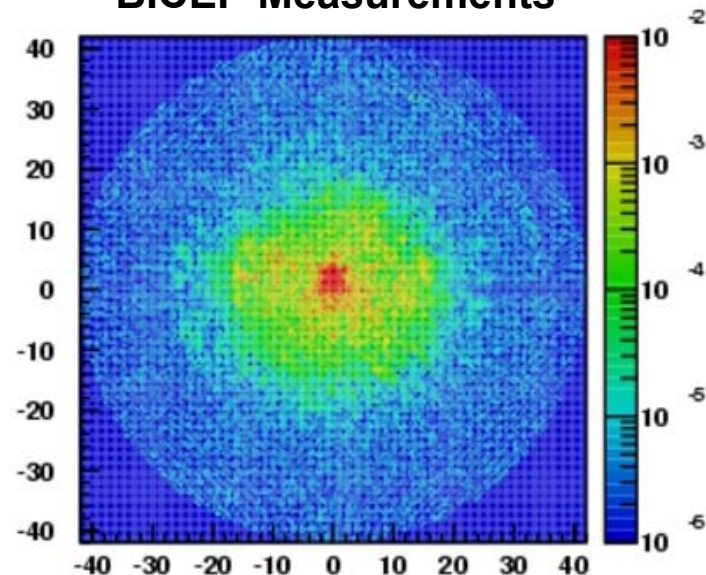
B-Polarization



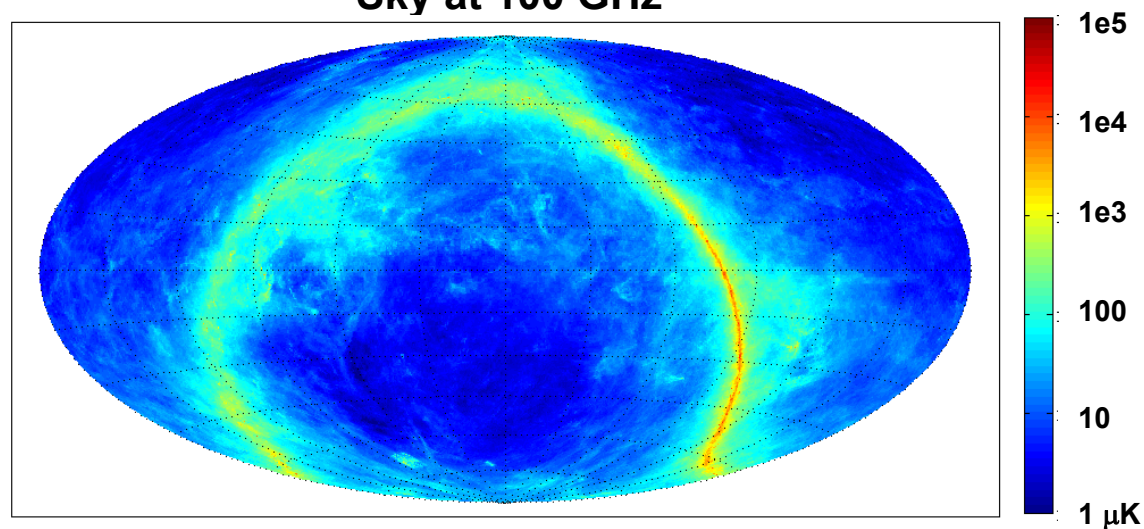


Far-Sidelobe Performance

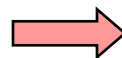
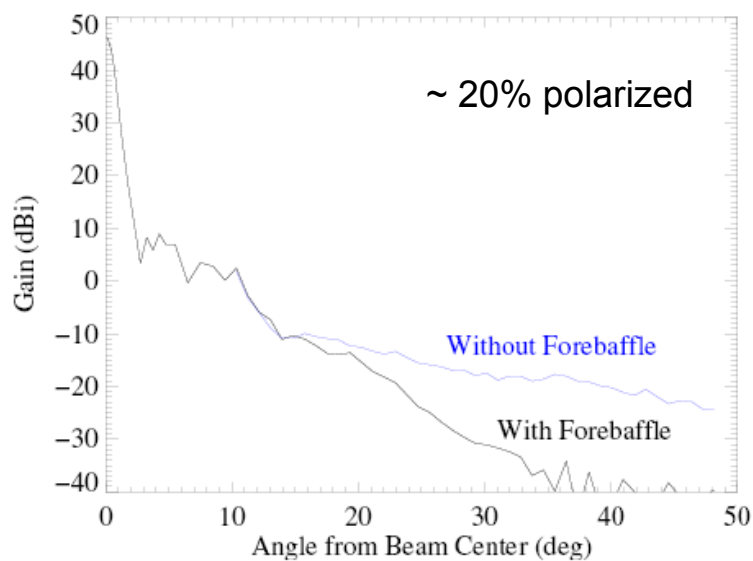
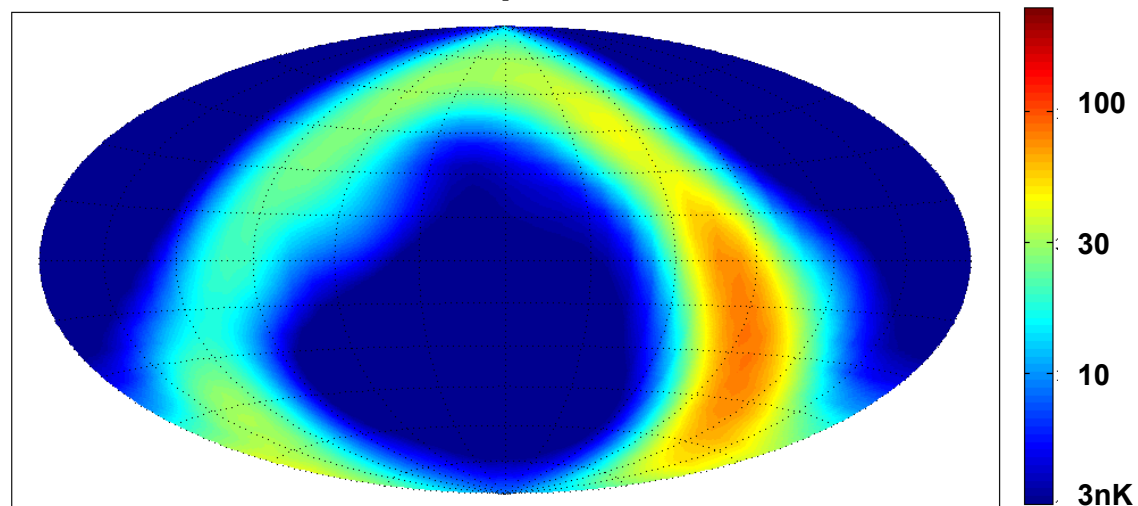
BICEP Measurements



Sky at 100 GHz



Sidelobe Map at 100 GHz



Levels below 3 nK for most of the sky!



Conclusions

CMB is the most powerful tool in cosmology today

- § **Planck will still leave the most exciting CMB observable unexplored**
- § **Clear role for a space-borne polarization mission**

Imaging polarimeter approach

- § **Technically and scientifically feasible**
- § **Simple technique. Established history.**
- § **EPIC is high-TRL, and meets *TFCR* requirements for IGW science**
- § **Extensive control of systematics designed in from the beginning**

Foregrounds

- § **What we know today about polarized foreground shows they can be subtracted below $r = 0.01$**

Technology Readiness

- § **We can build a very capable mission *today* with *existing technology***
- § **New technologies improve the mission & reduce technical risk**



Backup Materials from EPIC Mission Study Report



Science Requirements Flowdown

NASA Objective	EPIC Objective	Measurement Criteria	Instrument Criteria
Discover what powered the Big Bang... search for gravitational waves from the earliest moments of the Big Bang Discover the origin, structure, evolution, and destiny of the universe (NASA 2006 Strategic Plan)	Test Inflationary paradigm at GUT energy scales by probing Inflationary Gravitational Wave B-mode polarization signal to $r = 0.01$.	Detect BB signal at $r = 0.01^*$ <i>after foreground removal</i>	High sensitivity
			30 – 300 GHz [†]
		Positively detect both the $\ell = 5$ and $\ell = 100$ BB peaks	Control systematics to negligible levels
			All-sky coverage
			Low angular resolution ($< 1^\circ$)
Understand how the first stars and galaxies formed	Distinguish models of reionization history	Measure EE to cosmic variance	Parameters above
Determine the size, shape, and matter-energy content of the Universe	Extract all available EE cosmology		
Measure the cosmic evolution of the dark energy, which controls the destiny of the universe	Measure lensing BB to determine neutrino mass and dark energy equation of state	Measure lensing BB to $\sim$ cosmic variance	
	Remove lensing BB using shear map		
...Trace the flows of energy and magnetic fields... between stars, dust, and gas	Map Galactic magnetic fields	Measure synch and dust polarization	

- Primary Objective**
- Secondary Objective**

*Sensitivity goal stated by Weiss Committee TFCR

[†]Frequency range stated by Weiss Committee TFCR



EPIC Mission Requirements and Goals

Instrument Criteria	Requirements	Design Goals
High sensitivity	NET = 2 $\mu\text{K}\sqrt{\text{s}}$	NET = 1 $\mu\text{K}\sqrt{\text{s}}$
Subtract foreground signals to negligible levels	Technological capability to cover 30 – 300 GHz. Remove foregrounds to below $r = 0.01$ science goal	Optimize bands for foreground removal based on best knowledge
Control systematic errors to negligible levels	Suppress systematic errors to < 10 % of $r = 0.01$ signal, after correction 1-year required life for 2 complete maps for error checks	Suppress raw systematic effects to < 10 % of statistical noise level 2-year design life for multiple error checks at higher sensitivity
Maintain sensitivity on large angular scales	All-sky coverage with redundant interleaved scan strategy	
Angular resolution	1° at 100 GHz	



EPIC Mission Parameters

Mission Parameters		
Orbit	L2 Halo	
Mission Life	1-year (Required)	2-year (Design)
Bands	40 – 300 GHz	
Sensitivity	$2 \mu\text{K}_{\text{CMB}} \sqrt{\text{s}}$ (Required)	$1.5 \mu\text{K}_{\text{CMB}} \sqrt{\text{s}}$ (Design)
Resolution	16 – 155 arcmin (FWHM)	
Focal Plane	Antenna-Coupled NTD Bolometers	
Readout	Si JFETs mounted at 40 K	
Pol. Modulation	Half-wave Plate before Telescope Stepped 45° every 24 hours	
Optics	Six 30-cm Wide-Field Refractors	
Cryogenics	Passive to 40 K / LHe Cryostat to 2 K / Continuous ADR to 0.1 K	



Focal Plane Options

Input Assumptions

Fractional bandwidth $\Delta\nu/\nu = 30\%$

Focal plane temperature = 100 mK

Waveplate temperature = 20 K, with 2 % coupling

$P_{\text{sat}}/Q = 5$ for TES bolometers

Optical efficiency $\eta = 40\%$

Optics temperature = 2 K, with 10 % coupling

Baffle at 40 K with 0.3 % coupling (measured)

$G_0 = 10 Q / T_0$ for NTD bolometers

NTD Bolometer Option

Freq [GHz]	θ_{FWHM} [']	Nbol ³ [#]	Required Sensitivity ¹				Design Sensitivity ²			
			NET ⁴ [μK√s]		δT-θ ⁵ [μK ']	δTpix ⁶ [nK]	NET ⁴ [μK√s]		δT-θ ⁵ [μK ']	δTpix ⁶ [nK]
			bolo	band			bolo	band		
30	155	8	98	34.6	106	630	69	24.5	53.1	315
40	116	54	85	11.5	35.4	210	60	8.2	17.7	105
60	77	128	70	6.2	18.9	110	49	4.4	9.5	56
90	52	256	59	3.7	11.3	67	42	2.6	5.6	34
135	34	256	53	3.3	10.2	61	38	2.4	5.1	30
200	23	64	58	7.2	22.1	130	41	5.1	11.0	66
300	16	64	135	16.7	51.4	310	95	11.8	25.7	150
Total ⁷		830		2.1	6.5	39		1.5	3.3	19

TES Bolometer Option

Freq [GHz]	θ_{FWHM} [']	Nbol ³ [#]	Required Sensitivity ¹				Design Sensitivity ²			
			NET ⁴ [μK√s]		δT-θ ⁵ [μK ']	δTpix ⁶ [nK]	NET ⁴ [μK√s]		δT-θ ⁵ [μK ']	δTpix ⁶ [nK]
			bolo	band			bolo	band		
30	155	8	87	30.8	66.7	560	62	22	33.4	280
40	116	54	77	10.4	22.7	190	54	7.4	11.3	95
60	77	128	66	5.8	12.7	107	47	4.1	6.3	53
90	52	512	59	2.6	5.6	47	41	1.8	2.8	24
135	34	512	59	2.6	5.7	48	42	1.9	2.8	24
200	23	576	72	3.0	6.5	55	51	2.1	3.2	27
300	16	576	145	6.0	13.0	110	100	4.2	6.5	55
Total ⁷		2366		1.5	3.2	27		1.0	1.6	13

¹Sensitivity with $\sqrt{2}$ noise margin in a 1-year mission

²Calculated sensitivity in 2-year design life

³Two bolometers per focal plane pixel

⁴Sensitivity of one bolometer in a focal plane pixel

⁵Sensitivity δT in a pixel $\theta_{\text{FWHM}} \times \theta_{\text{FWHM}}$ times θ_{FWHM}

⁶Sensitivity δT in a $120' \times 120'$ pixel

⁷Combining all bands together



Risk Mitigation Strategy

Low Cost Mission Option Design Approach		
Instrument Requirements	Mitigation Approach	Mitigations
Sensitivity	NTD Ge detectors	Requirement includes $\sqrt{2}$ noise margin Upscope to TES bolometers when fully demonstrated
Subtract foreground signals below $r = 0.01$	Antenna-coupled bolometers	Technology covers full 30 – 300 GHz spectral band
	Single-color refracting telescopes	Optimized AR coatings and waveplate modulation for operation over full 30 – 300 GHz spectral band
Suppress systematic errors to < 10 % of $r = 0.01$ signal, after correction	Spinning/precessing scans	Uniform angular coverage over entire sky
	Highly redundant scan strategy, Step waveplate every 24 hours	Redundant daily maps with each pixel of 50 % of sky allow comprehensive jackknife tests.
		Immunity to data interruptions, bad pixels, bad arrays
		Two full maps in 1-year for systematic error testing
	Dual-polarization detector	Upscope to continuous waveplate with fully demonstrated
		Suppresses common-mode temperature signals, thermal drifts
	Waveplate modulator in front of telescope	Suppresses main beam systematics by rotating polarization without altering beam shapes
		Suppresses 1/f noise by signal modulation if continuous
		Suppresses gain and temperature drifts if continuous
	Small aperture monochromatic refracting telescope	Large throughput
		Low instrument- and cross- polarization
		Low main beam asymmetries
		Optimized low-reflection coatings
		Low far-sidelobe response
All-sky coverage	1-year required lifetime	Cryostat spec has 100 % margin
Minimize cost and risk	30 cm monochromatic refracting telescope	Low-cost polyethylene lenses with simple AR coatings Already demonstrated in BICEP
	LHe cryostat	Low technology risk
		Low integration risk - no microphonic, EMI or B-field disturbances
		Readily allows systems-level testing
	Commercial spacecraft	Low-cost standardized spacecraft with modifications
	Fixed downlink antenna	Eliminates risk of gimbal antenna



Mass Summary

Mass Summary: Low-Cost Option				
Sub-Assembly		Mass (CBE) [kg]	Contingency [%]	Allocated Mass [kg]
Focal Planes	Mass at 0.1 K per unit	0.9	35	1.2
	Mass at 0.4 K per unit	1.0	35	1.4
	Mass at 2 K per unit	0.5	35	0.7
	Total Focal Plane Assemblies (6)	14.2	35	19.2
Tele- scopes	Lenses at 2 K per unit	2.1	35	2.8
	Supports per unit	1.6	35	2.2
	Shields per unit	0.9	35	1.2
	Total Telescope Assemblies (6)	27.8	35	37.5
Wave- Plates	Waveplate 3-stack ave per unit	3.0	35	4.0
	Suspended bearing/motor per unit	1.5	35	2.0
	Non-suspended mass	2.5	35	3.4
	Total Waveplates (6)	41.7	35	56.3
Adiabatic Demagnetization Refrigerator		5.7	35	7.7
Ejectable Telescope Covers (6)		6.0	35	8.1
Cryostat and Shell	Liquid Helium	62.9	0	62.9
	Helium Tank	29.5	35	39.8
	Vapor-Cooled Shields	57.5	35	77.6
	Vacuum shell	185.0	35	249.8
	MLI	12.7	35	17.1
	Fill/vent lines, valves, ports	16.0	35	21.6
	Total Cryostat and Shell	363.6		468.8
Cabling		7.0	35	9.4
Warm Electronics		40.0	35	54.0
V-groove Radiators		51.3	35	69.3
Deployed Sunshield		74.1	35	100.0
Struts from S/C to Instrument		15.5	35	20.9
Spacecraft (based on SA200HP bus)		354.0	15	407.1
X-band Antenna and Transmitter		12.7	35	17.1
Propellant [$\Delta V = 160$ m/s]		122.0	0	122.0
Total		1149		1398
Launch Vehicle Maximum Payload Mass to L2 (C3 = -0.6)				
Vehicle		Mass	Margin [%]	Margin [kg]
Delta 2925H-9.5 Star 48		1440	3	42
Delta IV 4040		2773	98	1375
Atlas V 401		3485	150	2087



Power Summary

Power Summary: Low-Cost Option			
Item	Power (CBE) [W]	Contingency [%]	Allocated [W]
X-band TWTA	200	30	260
ADR Electronics	40	30	50
Bolometer Readout Electronics	150	30	195
Spacecraft	355	30	460
Total	935	30	965
GaAs Triple Junction Solar Panels			
Panel Area	Power [W]	Margin [%]	Margin [W]
4.0 m ² Fixed at 45° Incidence	710		
3.8 m ² Deployed at 45° Incidence	670		
Total	1380	43	415



Telemetry Budget

Antenna BW [deg]	Antenna Gain [dBi]	SC-Amp [W]	Gnd Station [m]	Gnd Gain [dBi]	Downlink Rate [Mbps]
2	9.0	100	12	58.65	0.5
2	9.0	90	34	68.29	4

Option	Spin rate [rpm]	Waveplate rate	Input Rate ¹ [kbps]	Downlink time per day [hrs]	
				12-m GS	34-m GS
Scan-modulated NTD bolometers ²	1.0	1 rev / day	87	4.2	0.5
Waveplate-modulated TES bolos ³	0.1	40 – 300 rpm	480	-	2.8
Scan-modulated TES bolometers ⁴	3.0	1 rev / day	1260	-	7.4

¹Assumes 4 bits per sample per detector (Planck compression ratio), and 2x Nyquist sampling.

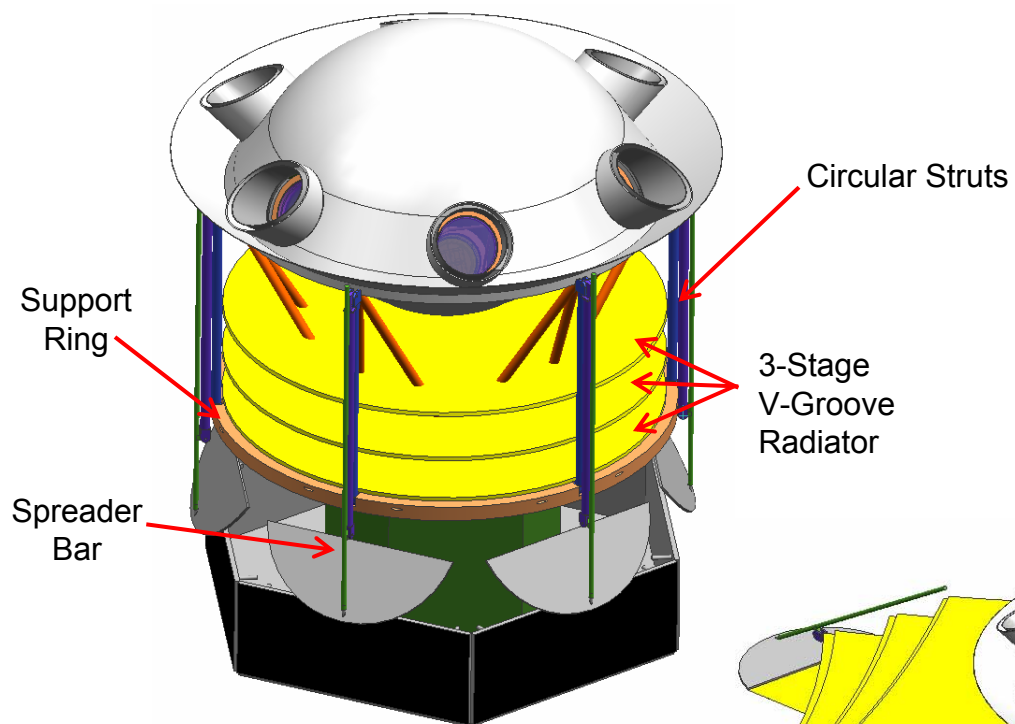
²Requires a 1/f knee < 16 mHz (already demonstrated for NTD bolometers).

³Assumes 10 polarization cycles per beam crossing for each band. Requires 1/f knee < 2.5 Hz.

⁴Requires a 1/f knee < 50 mHz (near state-of-the-art for TES bolometers)

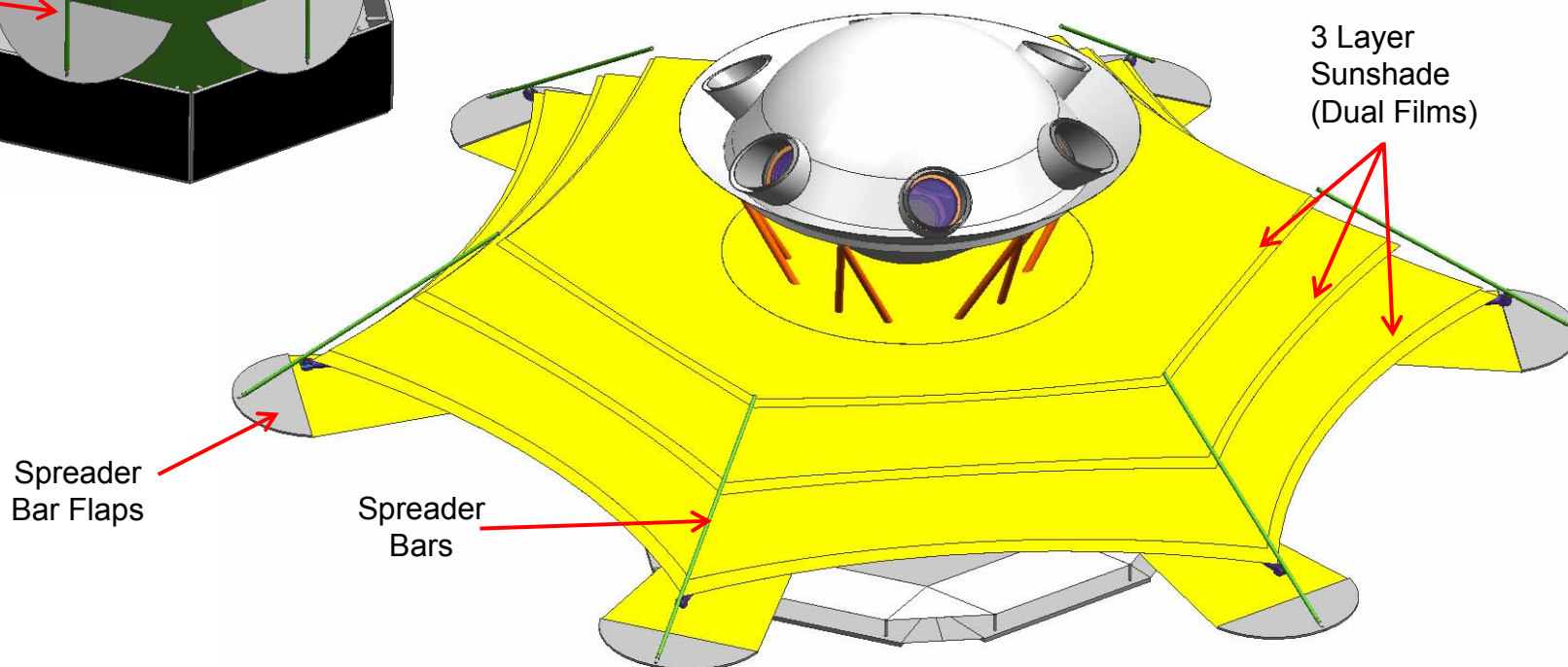


Deployable Sunshield Design (1)



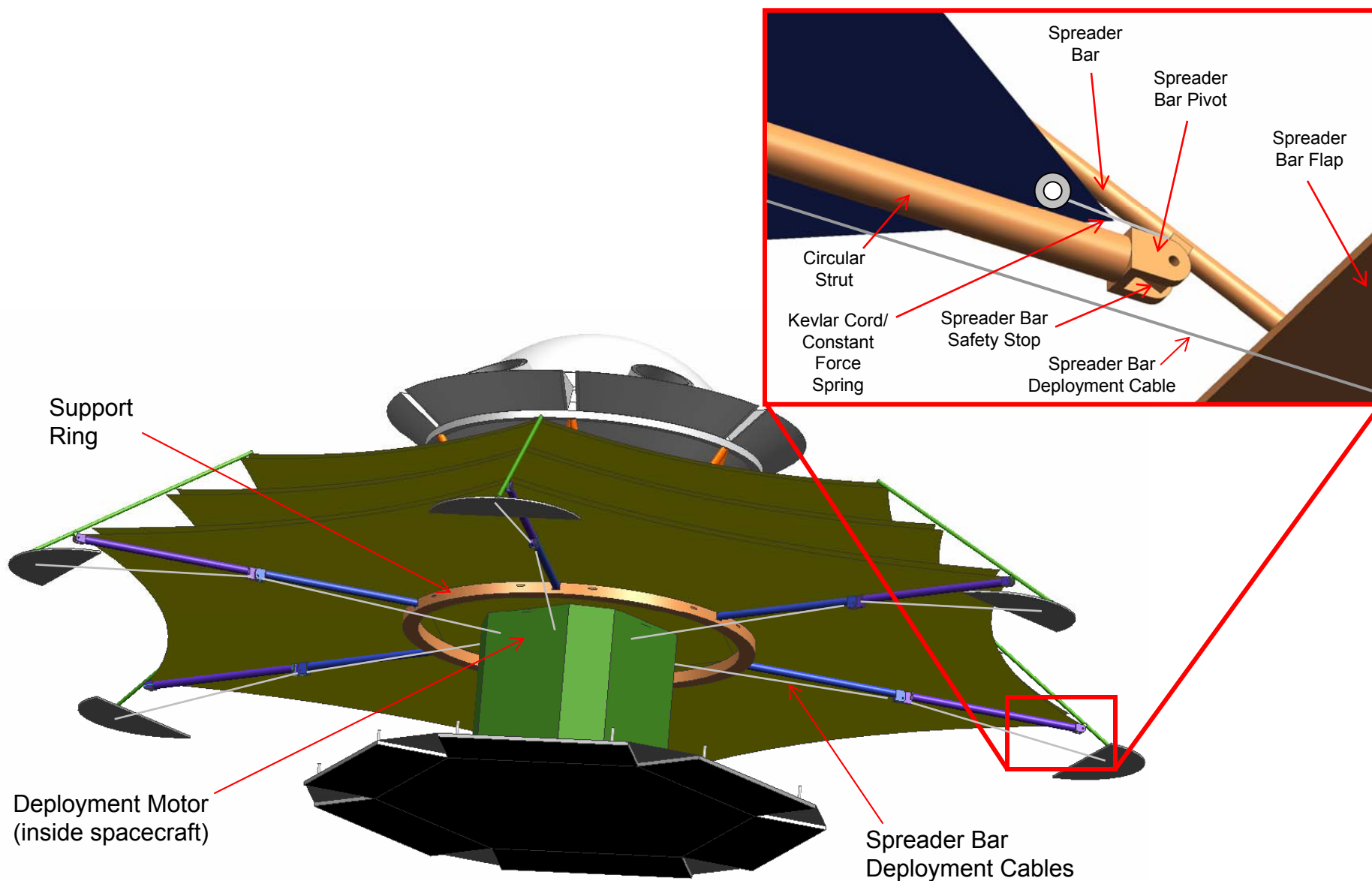
Sunshade Parameters

Shade largest diameter	8 m
Fairing	Delta II 2925 3-m
Fundamental frequency	> 0.17 Hz
Strut safety factor	> 6
Spreader bar safety factor	> 3
Sunshield layers	3, double layered
Sunshade and deployment hardware mass	74 kg



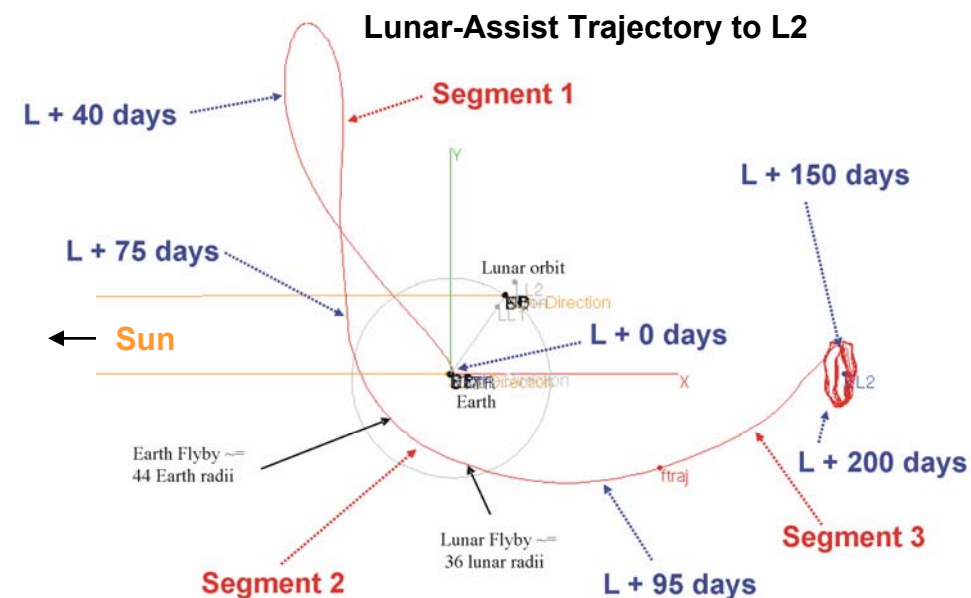


Deployable Sunshield Design (2)





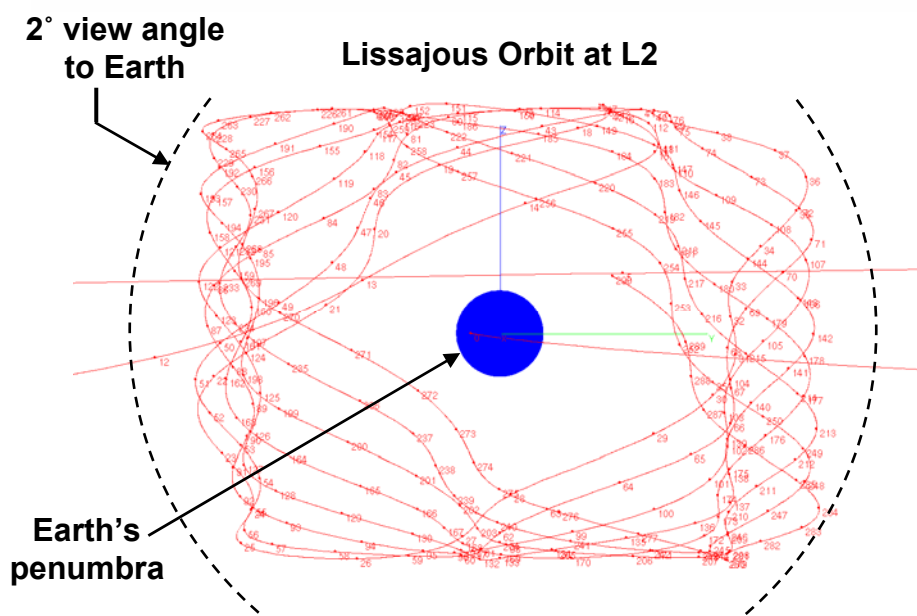
L2 Halo Orbit Parameters



Nominal Orbital Parameters	
Transfer Time	~170 Days
Halo Dimensions	24,000 × 46,000 × 33,000 km
Max Earth-view Angle	2°
Partial Eclipses	None over 4 years
Lunar Fly-by Distance	64,000 km
C3	-0.45 km ² /s ²
Deterministic ΔV	0 m/s

Summary of EPIC ΔV Budget (4 years)

Event / Function	Very Conservative		Semi Conservative	
	Mean	Sigma	Mean	Sigma
Injection Cleanup	64	45	58	34
Post-Injection up to Lissajous Insertion (L + 200 days)	7	3	7	3
Sub Total	72	45	65	34
Lissajous per Rev	4	1	4	1
4-Years Sub Total	31	10	31	10
Grand Total	103	46	96	35
Total 95% Probability	195		166	
Margin	10		10	
Total Budget	215		176	



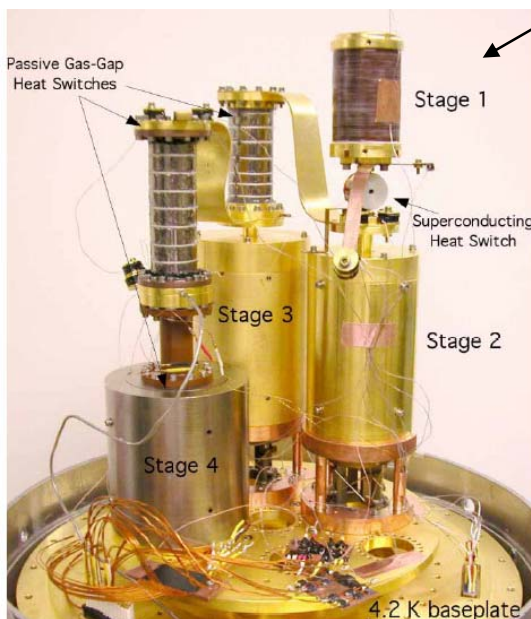
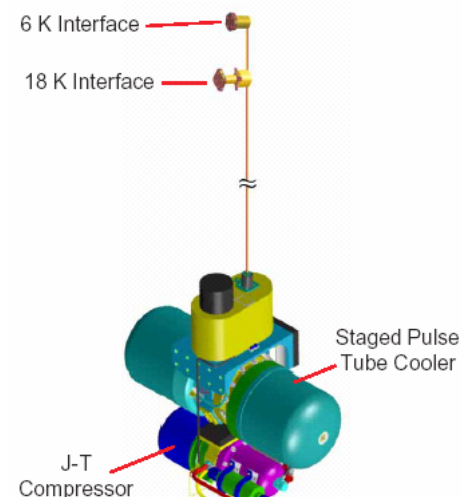


Available Cooler Options to 100 mK

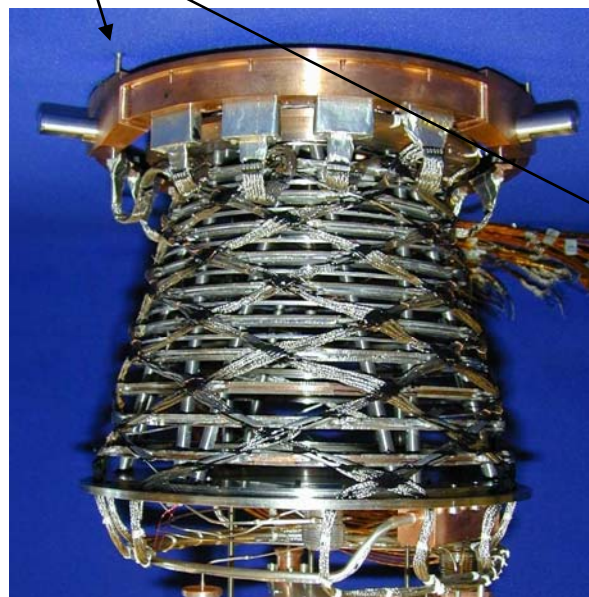
High-TRL Baseline Approach

1. **Passive cooling to 40 K**
- Temperatures are extremely stable
2. **Simple 450 ℓ liquid helium cryostat to 2 K**
3. **Single-shot ADR to 100 mK**
- We can tolerate data interruptions

Other Available Options

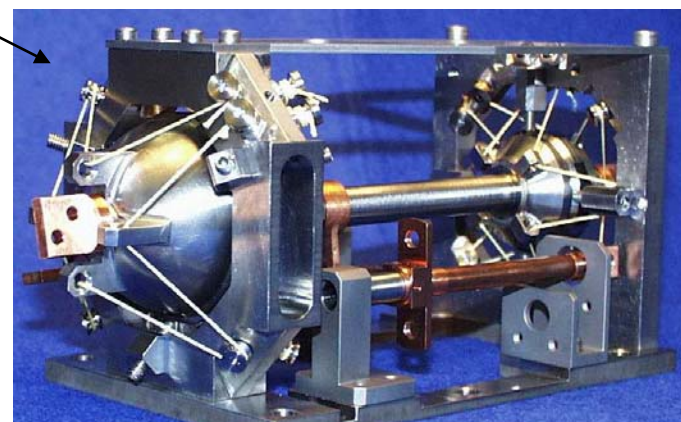


Continuous ADR
+ Continuous, low mass
- Low-TRL, T stability



Continuous OC Dilution Fridge
+ Flight-qualified, highly stable
- Low cooling power

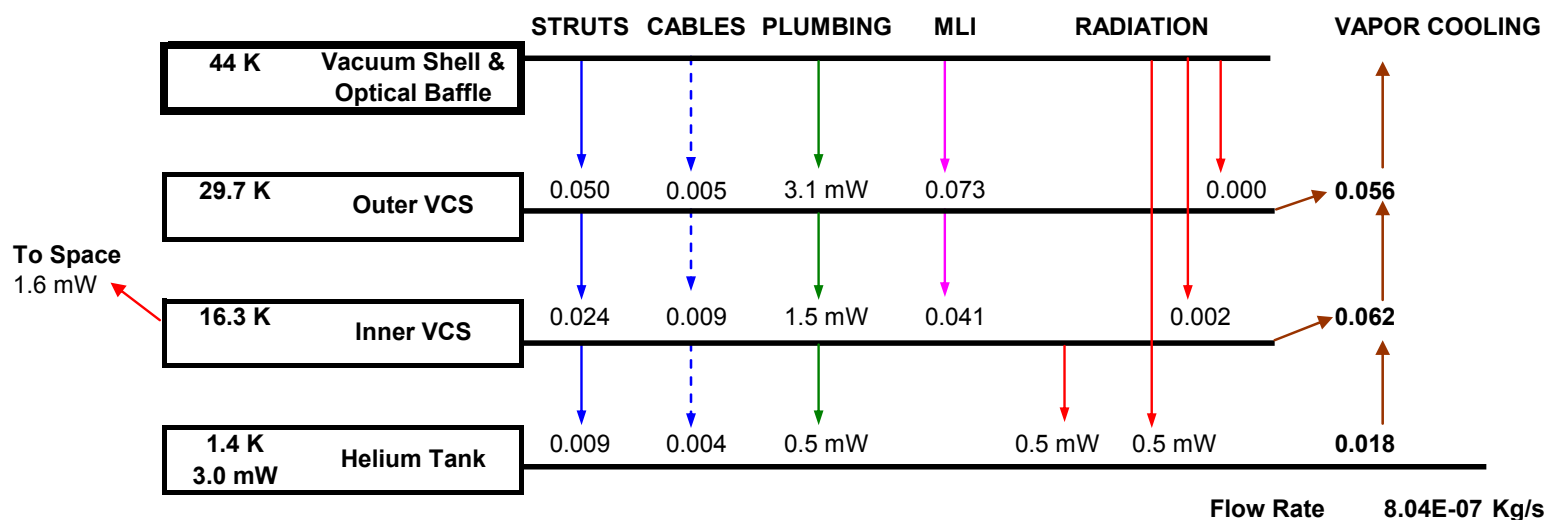
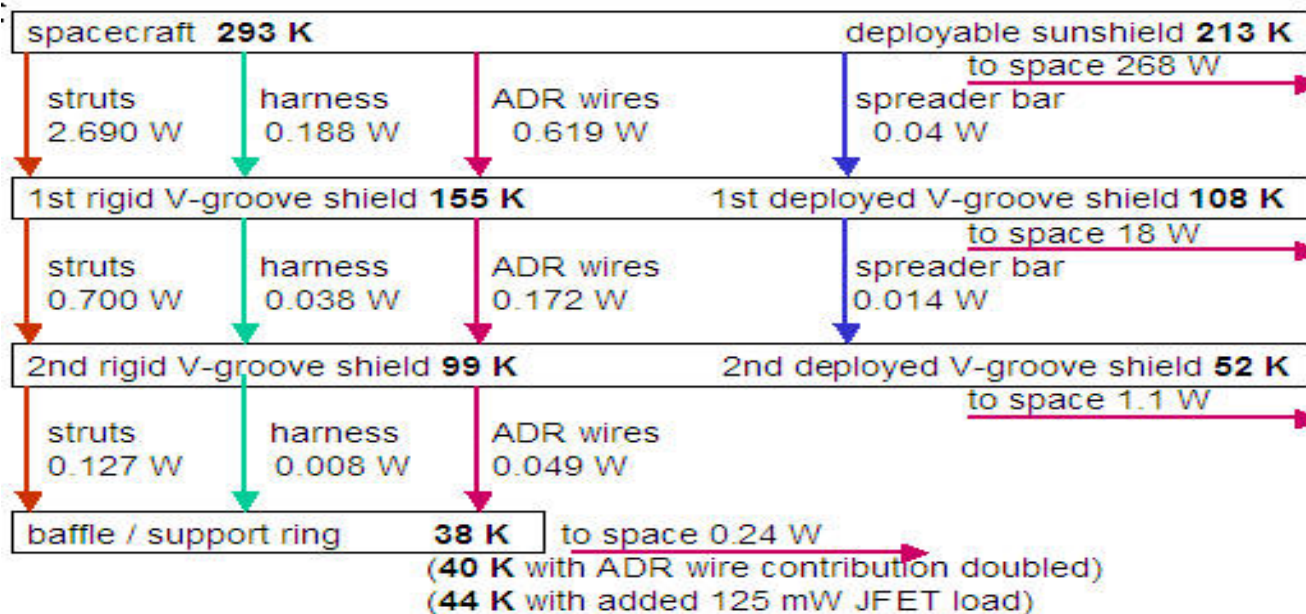
Mechanical Coolers to 2-4 K
+ Long life, low mass, JWST dev.
- T stability, μ -phonics, B-fields?



³He Sorption Cooler
+ Flight-proven, highly stable
- 300 mK, not continuous



Thermal Heat Flow Analysis



30.2 Month Primary Cryogen Lifetime
433 liter Primary Cryogen tank

Heat flow in Watts unless noted



Toroidal Beam Antenna

